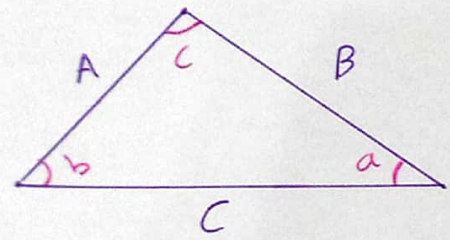


تلخيص ميكانيكا هندسية

للطالب المبدع
محمود مجدلاني

إرادة - ثقة - تغيير

Chapter 2

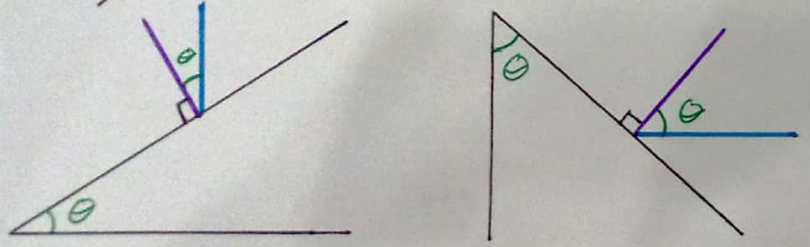
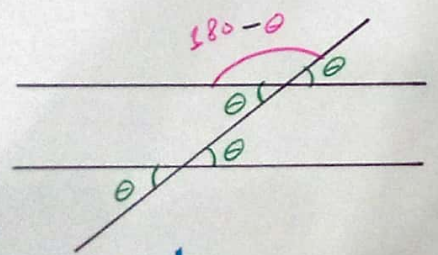


Cosine law

$$C = \sqrt{A^2 + B^2 - 2AB \cos \gamma}$$

Sine law

$$\frac{A}{\sin(\alpha)} = \frac{B}{\sin(\beta)} = \frac{C}{\sin(\gamma)}$$



$$\left. \begin{aligned} \cos \alpha &= \frac{A_x}{|A|} \\ \cos \beta &= \frac{A_y}{|A|} \\ \cos \gamma &= \frac{A_z}{|A|} \end{aligned} \right\}$$

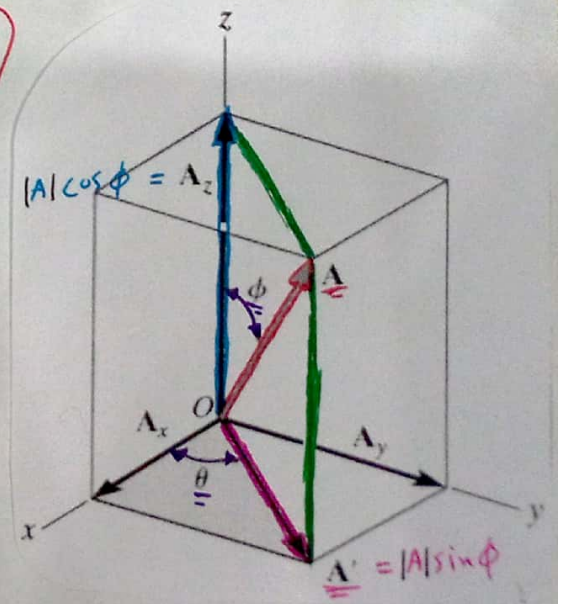
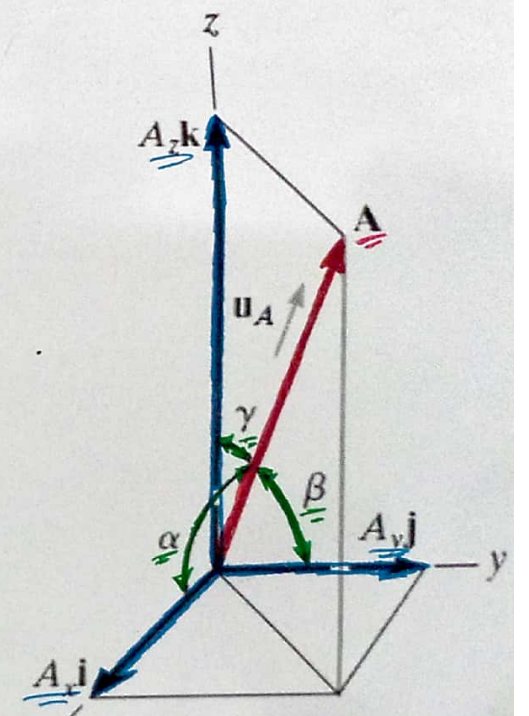
unit vector

$$\left. \begin{aligned} \frac{\vec{A}}{|A|} &\rightarrow \frac{A_x}{|A|} \hat{i} + \frac{A_y}{|A|} \hat{j} + \frac{A_z}{|A|} \hat{k} \\ u_A &\rightarrow \cos \alpha \hat{i} + \cos \beta \hat{j} + \cos \gamma \hat{k} \end{aligned} \right\}$$

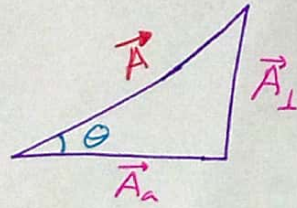
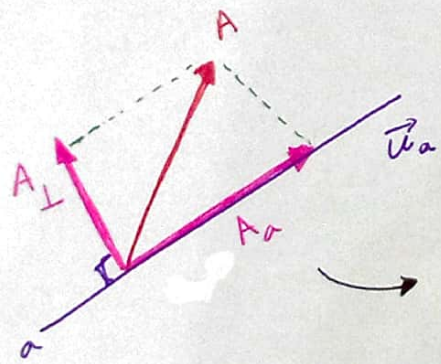
$$\left. \begin{aligned} \vec{A} &\rightarrow |A| u_A \\ &\rightarrow A \cos \alpha \hat{i} + A \cos \beta \hat{j} + A \cos \gamma \hat{k} \\ &\rightarrow A_x \hat{i} + A_y \hat{j} + A_z \hat{k} \end{aligned} \right\}$$

$$\cos^2 \alpha + \cos^2 \beta + \cos^2 \gamma = 1$$

$$\left. \begin{aligned} A' &= |A| \sin \phi \\ A_x &= A' \cos \theta \\ A_y &= A' \sin \theta \end{aligned} \right\}$$



Chapter 2

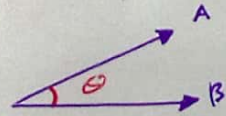


$$\diamond A_{\parallel} \begin{cases} A \cos \theta \\ \vec{A} \cdot \vec{u} \end{cases}$$

$$\begin{cases} |A_{\perp}| = \sqrt{A^2 - A_{\parallel}^2} \\ \vec{A}_{\perp} = \vec{A} - \vec{A}_{\parallel} \end{cases}$$

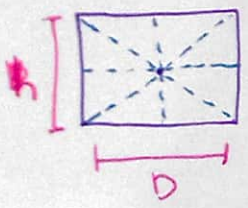
$$\diamond \theta = \cos^{-1} \left(\frac{A_{\parallel}}{A} \right)$$

$$\star \theta = \cos^{-1} \left(\frac{\vec{A} \cdot \vec{B}}{|\vec{A}| |\vec{B}|} \right)$$



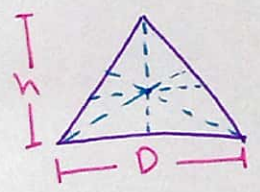
$$\hookrightarrow \underline{\underline{0 \leq \theta \leq 180}}$$

Chapter 9



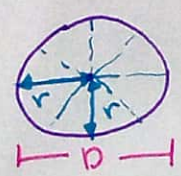
$$\bar{x} = \frac{D}{2}$$

$$\bar{y} = \frac{h}{2}$$

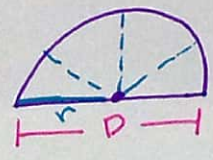


$$\bar{x} = \frac{D}{3}$$

$$\bar{y} = \frac{h}{3}$$

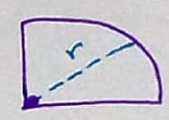


$$\bar{x} = \bar{y} = r$$



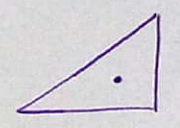
$$\bar{x} = r$$

$$\bar{y} = \frac{4r}{3\pi}$$



$$\bar{x} = \frac{4r}{3\pi}$$

$$\bar{y} = \frac{4r}{3\pi}$$



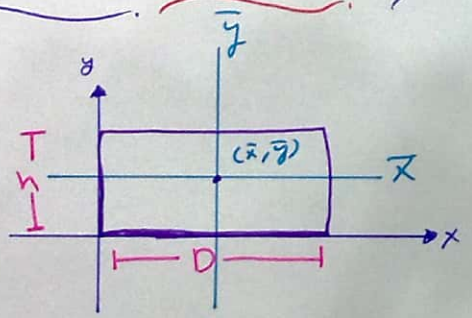
$$\bar{x} = \frac{b}{3}$$

$$\bar{y} = \frac{h}{3}$$

$$\bar{x} = \frac{\sum \bar{x}A}{\sum A}$$

$$\bar{y} = \frac{\sum \bar{y}A}{\sum A}$$

$$\bar{z} = \frac{\sum \bar{z}A}{\sum A}$$

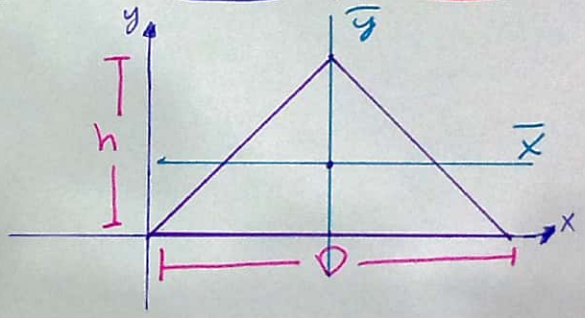


$$I_{\bar{x}} = \frac{Dh^3}{12}$$

$$I_x = \frac{Dh^3}{3}$$

$$I_{\bar{y}} = \frac{hD^3}{12}$$

$$I_y = \frac{hD^3}{3}$$



$$I_{\bar{x}} = \frac{Dh^3}{36}$$

$$I_x = \frac{Dh^3}{12}$$

$$I_{\bar{y}} = \frac{hD^3}{36}$$

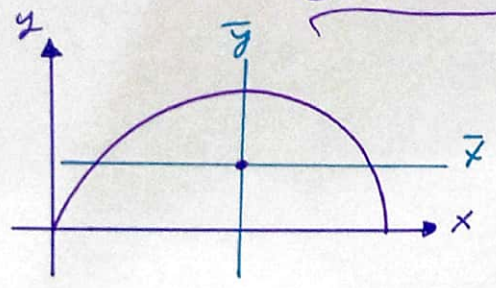
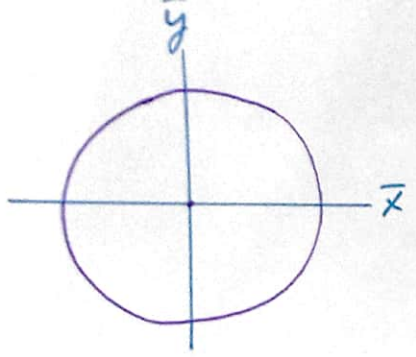
$$I_y = \frac{hD^3}{12}$$

Centroidal axis

Centroidal axis

$$I_x = I_{\bar{x}} + d^2 A$$

$$I_y = I_{\bar{y}} + d^2 A$$



$$I_{\bar{x}} = I_{\bar{y}} = \frac{\pi r^4}{4}$$

centroidsal axis

$$I_{\bar{x}} = \frac{\pi}{8} r^4$$

$$I_x = \frac{\pi r^4}{8} + \frac{8}{9\pi} r^4$$

Chapter (2-5)

Normal stress

$$\sigma_z = \lim_{\Delta A \rightarrow 0} \frac{\Delta F_z}{\Delta A}$$

Shear stress

$$\tau_{zx} = \lim_{\Delta A \rightarrow 0} \frac{\Delta F_x}{\Delta A}$$

$$\tau_{zy} = \lim_{\Delta A \rightarrow 0} \frac{\Delta F_z}{\Delta A}$$

Avg. Normal stress

$$\sigma = \frac{N_{max}}{A} \text{ [Pa]}$$

Avg. shear stress

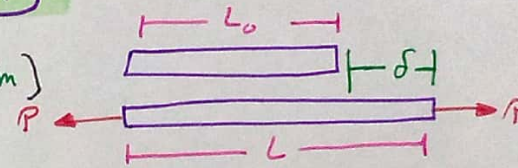
$$\tau_{avg} = \frac{V}{A} \text{ [Pa]}$$

Factor of safety

$$F.S = \frac{F_{fail}}{F_{allow}} = \frac{\sigma_{fail}}{\sigma_{allow}} = \frac{\tau_{fail}}{\tau_{allow}}$$

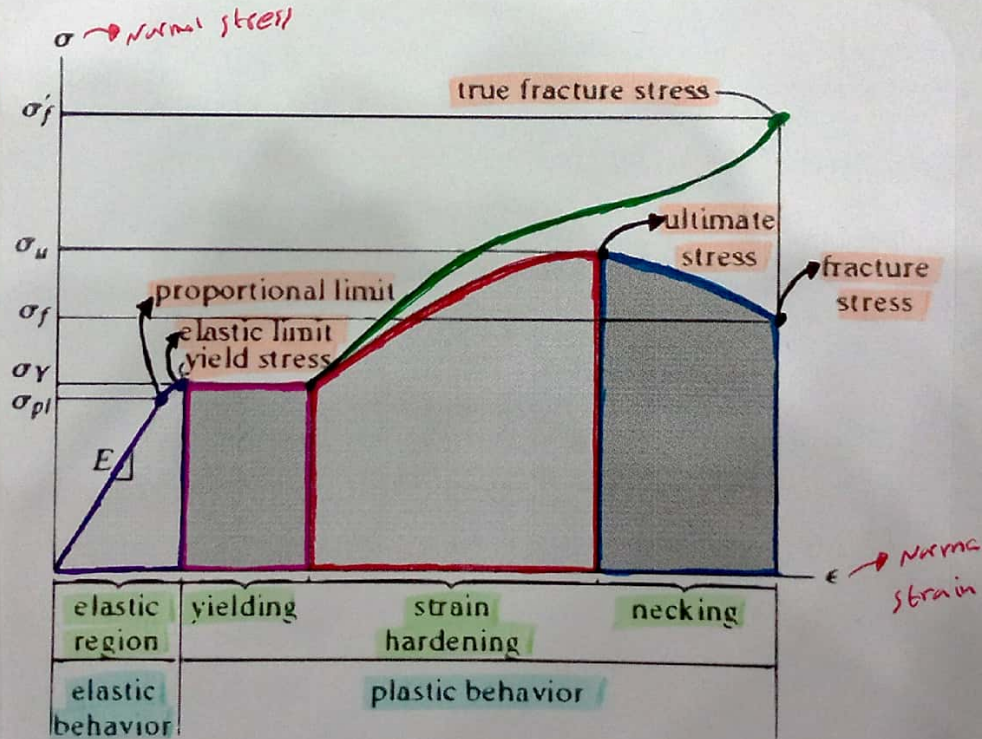
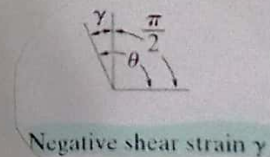
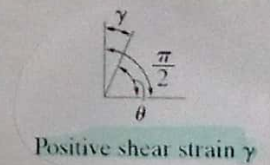
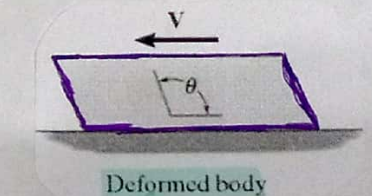
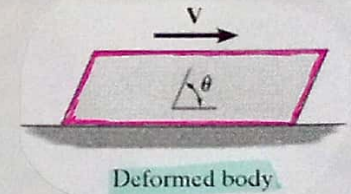
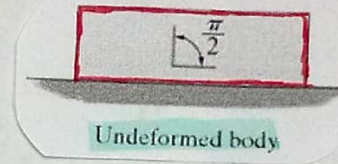
Avg. Normal strain

$$\epsilon_{avg} = \frac{L_{ren} - L_{old}}{L_{old}} \text{ [m/m]}$$



shear strain

$$\gamma = \frac{\pi}{2} - \theta$$



Conventional and true stress-strain diagram for ductile material (steel) (not to scale)

Chapter (1-5)

Conventional Stress-Strain Diagram

$$\sigma = \frac{P}{A_0}$$

يظهر من القوة من
الهندسة ليكون بذلك
Yielding point

Normal, Engineering strain

$$\epsilon = \frac{\delta}{L_0}$$

Hooke's law

$$\sigma = E \epsilon_{cong/PL}$$

Modulus of elasticity
Young modulus

$$E = \frac{\Delta \sigma}{\Delta \epsilon}$$

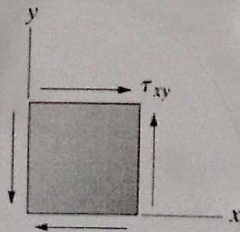
$$\tau = G \gamma$$

shear modulus of elasticity
modulus of rigidity

$$G = \frac{\tau_{RL}}{\gamma_{RL}}$$

في حالة القص

$$G = \frac{E}{2(1+\nu)}$$



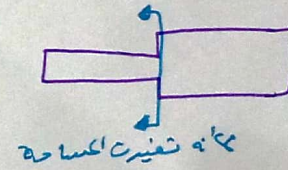
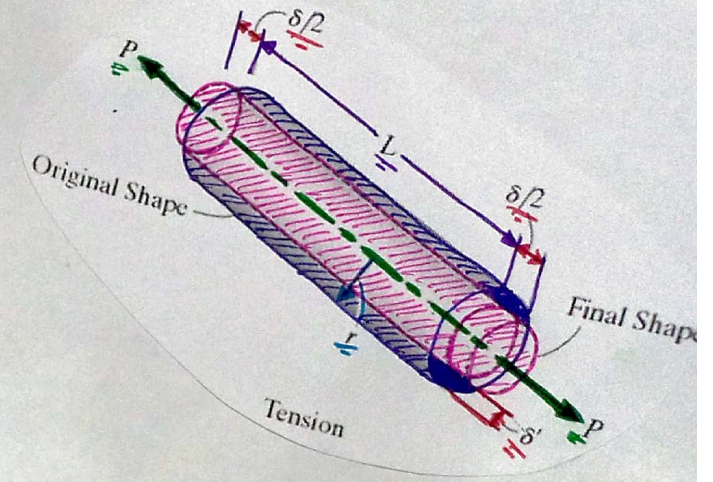
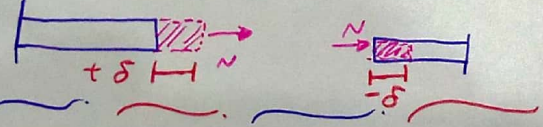
Poisson's ratio

$$\nu = -\frac{\epsilon_{lat}}{\epsilon_{long}}$$

$$\epsilon_{long} = \frac{\delta}{L}$$

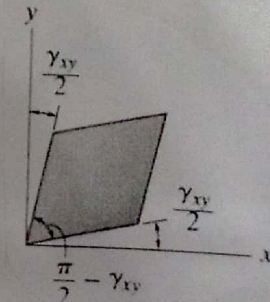
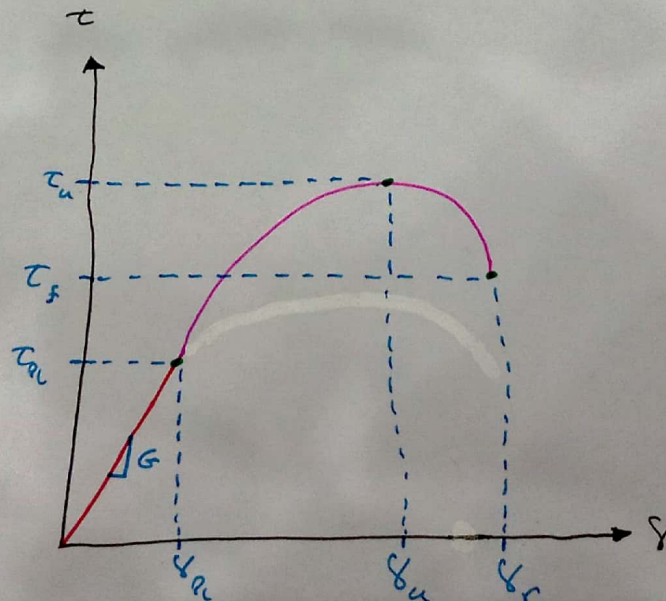
$$\epsilon_{lat} = \frac{\delta'}{r}$$

$$\delta = \sum \frac{NL}{AE}$$



تغير المساحة

* yielding = E = \sigma_y



Torsion formula

$$\tau_{max} = \frac{T}{J} c$$

$$\tau = \frac{T}{J} \rho$$

Polar moment of inertia



$$\phi = \left(\frac{\rho}{c}\right) \phi_{max}$$

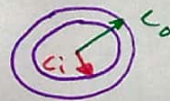
$$\tau = \left(\frac{\rho}{c}\right) \tau_{max}$$

Polar Moment of inertia

$$J_y = \frac{\pi}{2} C^4 \quad m^4$$

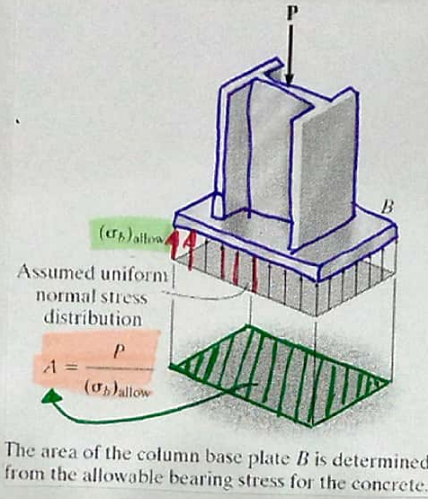
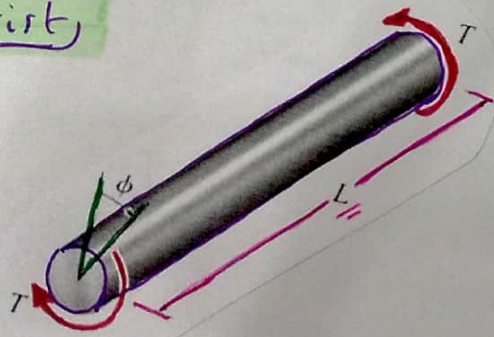


$$J_z = \frac{\pi}{2} (C_o^4 - C_i^4)$$

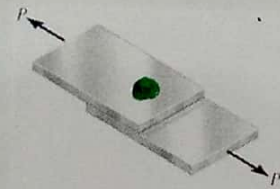


Angle of twist

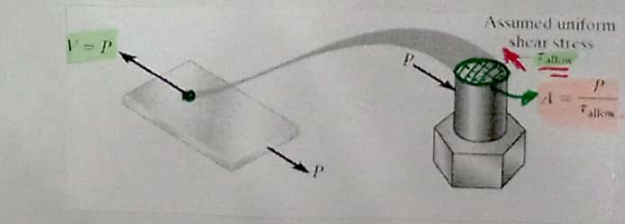
$$\phi = \sum \frac{TL}{JG}$$



$$A = \frac{P}{(\sigma_b)_{allow}}$$



The area of the bolt for this lap joint is determined from the shear stress, which is largest between the plates.



$$A = \frac{P}{\tau_{allow}}$$



$$L = \frac{P}{\pi d \tau_{allow}}$$

flexer formula

$$\sigma_{max} = \frac{MC}{I}$$

Normal stress at any distance

$$\sigma = - \frac{My}{I}$$

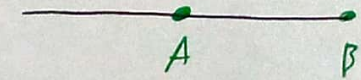
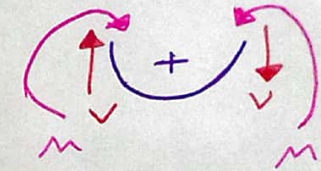
* Note

◆ $-\sigma \rightarrow$ Compression

◆ $+\sigma \rightarrow$ Tension

Shear formula

$$\tau = \frac{VQ}{It}$$



A \rightarrow إذا كانت الـ M
تقارب الساق

فيتها \oplus

B \rightarrow وانكبت فيتها \ominus