

تقدم لجنة EICoM الاكاديمية

تلخيص لمادة

# تحكم متقدم

من شرح:

د. سعد العجلوني

جزيل الشكر للطالبة:

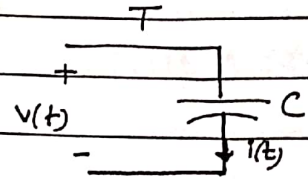
دانيا عمر



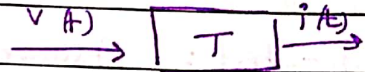
✓ Lec 1:

Review signals vs systems (LT I)

→ A signal: a function Ex  $(v(t))$



→ A system: A transformation



\* The capacitor system:  $T: v(t) \rightarrow i(t) = T[v(t)]$

$$i(t) = C \frac{dv(t)}{dt} \equiv C \lim_{\Delta t \rightarrow 0} \frac{v(t+\Delta t) - v(t)}{\Delta t}$$

\* A linear system satisfies: ① scaling property.  $T\{av_1(t)\} = ai_1(t)$   
 $\rightarrow T\{av_1(t)\} = ai_1(t)$   
 ② superposition property.

$T: v_1(t) \rightarrow i_1(t)$        $v_1(t) \rightarrow \boxed{\text{Linear } T} \rightarrow i_1(t)$

$T: v_2(t) \rightarrow i_2(t)$        $v_2(t) \rightarrow \boxed{\text{Linear } T} \rightarrow i_2(t)$

Linear  $T: v_1(t) + v_2(t) \rightarrow i_1(t) + i_2(t)$   
 $v_1 + v_2 \rightarrow \boxed{\text{Linear } T} \rightarrow i_1 + i_2$

Q: Is the system  $T_x: x(t) \rightarrow y(t) = 3x(t) + 1$

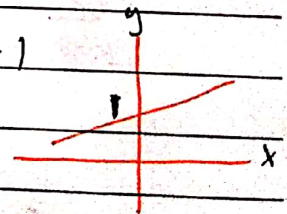
S.P  $\rightarrow T_x: x_1 \rightarrow 3x_1 + 1 = y_1$

$T_x: x_2 \rightarrow 3x_2 + 1 = y_2$

$T_x: x_1 + x_2 \stackrel{??}{=} y_1 + y_2$  ✗

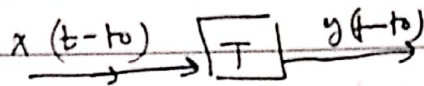
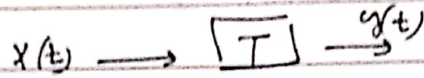
Scaling  $\rightarrow x_1 \rightarrow y_1 = 3x_1 + 1$

$5x_1 \rightarrow 3(5x_1) + 1 \neq 5(3x_1) + 1$  ✗



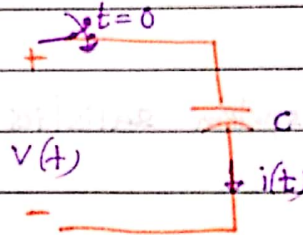
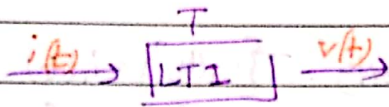
TI: if we delay an input before processing, output will be equal to output delayed after processing. [2]

TI systems has a time dependent function that is not affected by function of time.



TI-test: Assume input:  $x(t) \rightarrow y(t)$

If  $T\{x(t-t_0)\} = y(t-t_0)$ , then system T is "TI"



$$v(t) = \frac{1}{C} \int_0^t i(\tau) d\tau$$

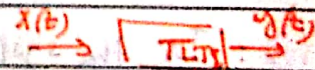
Lec 2:

Many physical processes (system) can be described as the following

DE:

$$\sum_{k=0}^m C_k \frac{d^k y(t)}{dt^k} = \sum_{k=0}^n b_k \frac{d^k x(t)}{dt^k}$$

assume:



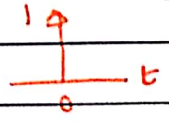
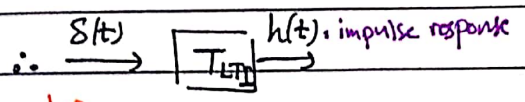
let  $m > n$

where  $C_k, b_k$  are constants

① such systems are "LTI" systems.

⊗ output of LTI systems due to an arbitrary input (convolution)

$$\begin{aligned}
 x(t) \xrightarrow{\text{LTI}} y(t) &= x(t) * h(t) = h(t) * x(t) \\
 &= \int_{-\infty}^{\infty} x(\tau) h(t-\tau) d\tau \\
 &= \int_{-\infty}^{\infty} h(\tau) x(t-\tau) d\tau
 \end{aligned}$$



⊗ If LTI is causal then  $\rightarrow h(t) = 0$  for  $t < 0$ .

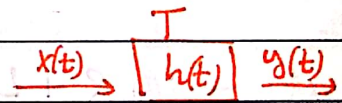
$$\rightarrow \int_0^{\infty} h(\tau) x(t-\tau) d\tau$$

⊗ It's hard to analyze a system using convolution.

→ Solution: Use freq domain.

⊗ Definition: Frequency Response  $\stackrel{\text{def}}{=} F\{h(t)\} = H(j\omega) = \int_{-\infty}^{\infty} h(t) e^{-j\omega t} dt$

→ If causal:  $H(j\omega) = \int_0^{\infty} h(t) e^{-j\omega t} dt$



Let  $s = j\omega \Rightarrow H(s) = \int_0^{\infty} h(t) e^{-st} dt = \mathcal{L}\{h(t)\} \Big|_{s=j\omega}$

⊗ Freq Response  $\equiv H(j\omega) \equiv F\{h(t)\} \equiv \mathcal{L}\{h(t)\} \Big|_{s=j\omega}$  ⊗

Why do analysis in freq Domain? :-

① Convolution becomes multiplication  
 in Time Domain  $\longleftrightarrow$  in Freq Domain

T.D

F.D

$$y(t) = h(t) * x(t) \xrightarrow{F} Y(j\omega) = H(j\omega) \cdot X(j\omega) \quad \text{Freq. Response}$$

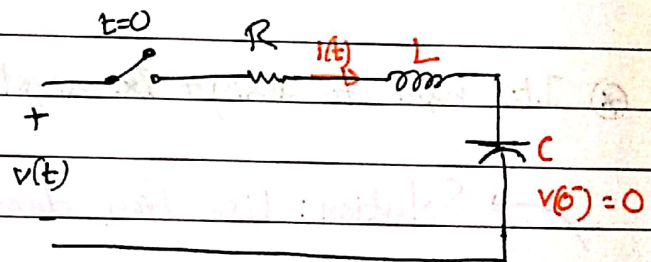
$$\xrightarrow{I} Y(s) = \underbrace{H(s)}_{T.F} \cdot X(s)$$

② DE's become Algebraic Equations

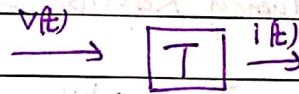
✓  $\rightarrow$  Table 2.1 and 2.2 pag. 36

✓  $\rightarrow$  Ex 2.1 and 2.2 in text book page: 37.

Ex RLC circuit:



find a math relationship between input  $v(t)$  and output  $i(t)$ ?



KVL  $v(t) = i(t)R + L \frac{di(t)}{dt} + \frac{1}{C} \int i(t) dt$

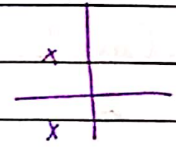
$\int$   $\left( \dot{v}(t) = \frac{di(t)}{dt} \cdot R + L \frac{d^2i(t)}{dt^2} + \frac{1}{C} i(t) \right)$  This is a LTI system then find T.F. For the

$S$   $V(s) = RS I(s) + S^2 I(s) + \frac{1}{C} I(s)$  system, i.e.,  $H(s) = \frac{I(s)}{V(s)}$

$$T.F = \frac{V(s)}{I(s)} = \frac{S}{LS^2 + RS + \frac{1}{C}}$$

lec 3:3

$\rightarrow sV(s) = RS I(s) + LS^2 I(s) + \frac{1}{C} I(s)$



$H(s) = \frac{I(s)}{V(s)} = \frac{s}{LS^2 + RS + \frac{1}{C}} = \frac{s}{(s+p_1)(s+p_2)}$   $\rightarrow p_1, p_2$  are complex numbers.

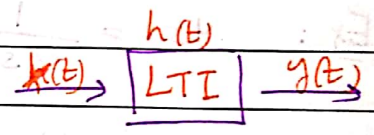
⊕ Find  $h(t) = \mathcal{L}^{-1}\{H(s)\}$

$\Downarrow$   
P.F.E  
 $= \frac{k_1}{s+p_1} + \frac{k_2}{s+p_2}$

$\mathcal{L}^{-1} \Downarrow$   
 $k_1 e^{-p_1 t} u(t)$

⊗ In general, how would you find the response (to a LTI sys) using inverse Laplace transform?

$Y(s) = H(s) X(s)$



steps to get  $y(t)$  from  $Y(s)$ :

① Rewrite  $Y(s)$  in the form of  $Y(s) = A(s) + \frac{N(s)}{D(s)}$ ;  $A(s), N(s), D(s)$  are polynomials in  $s$

$\rightarrow$  such that the degree of  $N(s)$  must be strictly less than the degree of  $D(s)$ .

② Conduct a partial fraction expansion (P.F.E) on  $\frac{N(s)}{D(s)}$

$\rightarrow$  P.F.E: let  $\frac{N(s)}{D(s)} = \frac{(s+z_1)(s+z_2)\dots(s+z_n)}{(s+p_1)(s+p_2)\dots(s+p_m)}$ ,  $M, \emptyset$   
 $z_i, p_j$  are complex.

$\therefore i=1, 2, \dots, m$   
 $j=1, 2, \dots, \emptyset$

Case 1: If all poles are distinct [each pole is of multiplicity = 1]

then  $\rightarrow \frac{N(s)}{D(s)} = \frac{k_1}{s+p_1} + \frac{k_2}{s+p_2} + \dots + \frac{k_q}{s+p_q}$

$k_j$  [j=1,2,...,q] are called "residues"

where  $R(j) = \left[ \frac{N(s)(s+p_j)}{D(s)} \right]_{s \rightarrow p_j}$

Case 2: If a pole,  $p_j$ , has multiplicity > 1

Ex:  $y(s) = \frac{1}{(s+1)(s+3)^2}$

(HW) page 40. ✓

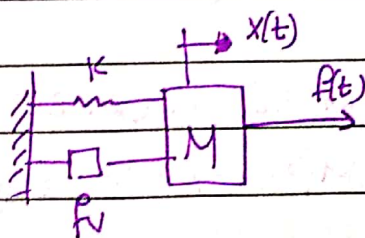
Case 1 Ex:  $y(s) = \frac{1}{s(s+1)}$  Find  $y(t)$ .

Sol: rewrite  $y(s) = \frac{A(s)}{s} + \frac{N(s)}{D(s)}$   $A(s)=0$   $N(s)=1$   $D(s)=s(s+1)$   
 $\rightarrow = (1 - e^{-t}) u(t)$

Section 2.5 translational mechanical sys:

table 2.4 and Ex 2.16, page 62-63.

Ex 2.16



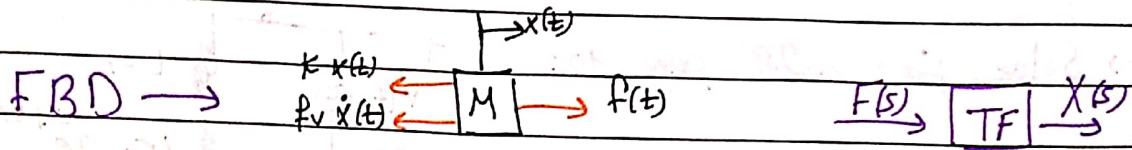
⊗ how to find EOM  $\xrightarrow{\text{Draw}}$  FBD

⊗ Assume the mass M is displaced in the +ve displacement direction



⊕ nominal point: system's initial point

steady state



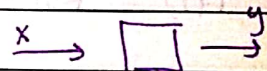
$$\sum F = M \frac{d^2 x(t)}{dt^2} = F(t) - F_v \dot{x}(t) - k x(t)$$

$$M\ddot{x} + F_v \dot{x} + kx = F(t), \text{ And the TF} = \frac{X(s)}{F(s)}$$

$$TF = \frac{X(s)}{F(s)} = \frac{1}{Ms^2 + F_v s + k}$$

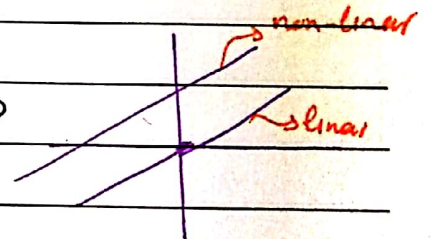
lec 4: Sec 2.10: Non linear systems

Sec 2.11: Linearization:



⊗ How to linearize a non-linear function, f?

$$F: X \rightarrow y \quad x \rightarrow [f] \rightarrow y$$



① linearization is done about a nominal point,  $x_0$ . open page

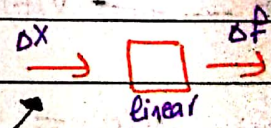
② expand f about  $x_0$  using Taylor Series expansion

$$\rightarrow \text{let } x = x_0 + \Delta x$$

$$f(x) = f(x_0 + \Delta x) = f(x_0) + \left. \frac{df(x)}{dx} \right|_{x=x_0} \Delta x + \frac{d^2 f(x)}{dx^2} \bigg|_{x=x_0} \frac{\Delta x^2}{2!} + \frac{d^3 f(x)}{dx^3} \bigg|_{x=x_0} \frac{\Delta x^3}{3!} + \dots$$

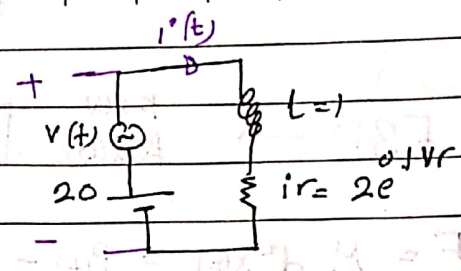
$$y = f(x) = f(x_0 + \Delta x) \approx f(x_0) + \left. \frac{df(x)}{dx} \right|_{x=x_0} \Delta x$$

$$\text{let } f(x) - f(x_0) = \Delta f = \left. \frac{df(x)}{dx} \right|_{x=x_0} \Delta x = m_{x_0} \Delta x$$





→ Solve Ex: 2.28 page 90:



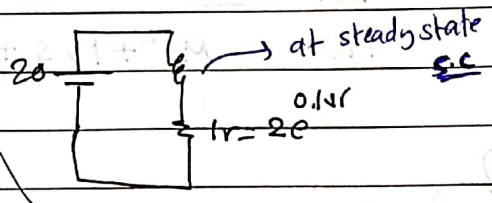
→ Find TF  $V_L(s)$  :-  
 $V(s)$

let  $v_0 = 20$  using KVL:

$$20 + v(t) = \frac{di}{dt} + 10 \ln \frac{i^2}{2} \quad \rightarrow \text{non-linear } P(i) \quad \text{[Initial goal: linearization about } i_0 \text{]}$$

$$\therefore i_0 = 2e^{0.1(20)} = 14.78$$

$$\rightarrow 10 \ln \frac{i_0}{2} = 20$$



let  $i = i_0 + \Delta i$

$$\hookrightarrow P(i) = P(i_0 + \Delta i) = 10 \ln \frac{i}{2} = P(i_0) + \left. \frac{dP}{di} \right|_{i=i_0} \Delta i$$

$$= 10 \ln \frac{i_0}{2} + 10 \left( \frac{1}{i_0} \times \frac{1}{2} \right) \Delta i$$

$$= 20 + 10 \frac{1}{10} \Delta i$$

⊗ Recall from KVL:  $20 + v(t) = \frac{di}{dt} + 20 + 0.67 \Delta i$

$$\frac{di(t)}{dt} = \frac{d(i_0 + \Delta i)}{dt} = \frac{di_0}{dt} + \frac{d\Delta i}{dt}$$

$$\text{Eq } \otimes \rightarrow v(t) = \frac{d\Delta i(t)}{dt} + 0.67 \Delta i \quad \rightarrow \text{let } \mathcal{L}\{\Delta i(t)\} = \Delta I(s)$$

taking  $\mathcal{L}$

$$V(s) = s \Delta I(s) + 0.67 \Delta I(s)$$

$$\Delta I(s) = \frac{V(s)}{s + 0.67}$$

Goal:  $v_L(t) = L \frac{di(t)}{dt}$

$$v_L(t) = L \frac{d\Delta i}{dt}$$

$$V_L(s) = sL \Delta I(s)$$

$$\rightarrow \frac{V_L(s)}{V(s)} = \frac{s}{s+0.67} \quad \#$$

Ch 3: Intro to modeling in the time-domain

(State-space (s-s) representation).

(HW) Study App. G (see moodle)



⊕ Given a LTI system represented by an  $n$ th order ODE; then the (s-s) representation is obtained by rewriting the ODE as follows:

$$\text{[state- eq]} \quad \dot{x}(t) = A x(t) + B u(t)$$

$\begin{matrix} n \times 1 & n \times n & n \times 1 & n \times m & m \times 1 \end{matrix}$

$$\text{[output- eq]} \quad y(t) = C x(t) + D u(t)$$

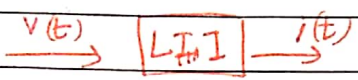
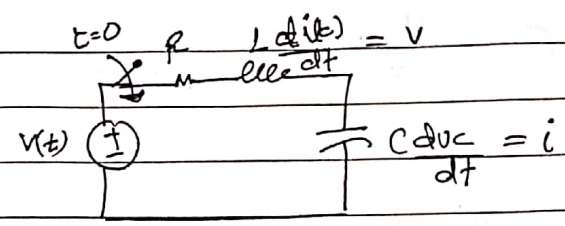
$\begin{matrix} p \times 1 & p \times n & n \times 1 & p \times m & m \times 1 \end{matrix}$

$A, B, C, D$  are constant matrices.

⊗ state variable:  $x(t)$  (var)   
 input:  $u(t)$

lec 5:

Ex RLC circuit:



(KVL) 
$$v(t) = i(t)R + L \frac{di(t)}{dt} + \frac{1}{C} \int_0^t i(z) dz$$

$x_1(t) = i(t)$   
 $x_2(t) = v_c(t)$

Recall:  $i(t) \rightarrow \frac{dq}{dt} \Rightarrow \int i(z) dz = q(t)$

$$v(t) = \dot{q}R + L\ddot{q} + \frac{1}{C}q$$

let:  $x_1(t) = q(t)$   
 $x_2(t) = \dot{q}(t) = \dot{x}_1(t) \Rightarrow \dot{x}_1(t) = x_2$

$$\dot{x}_2(t) = \frac{-1}{LC}x_1(t) - \frac{R}{L}x_2(t) + \frac{1}{L}v(t)$$

let 
$$x(t) = \begin{bmatrix} x_1 \\ x_2 \end{bmatrix}$$

$$\dot{x} = \begin{bmatrix} \dot{x}_1(t) \\ \dot{x}_2(t) \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ \frac{1}{LC} & -\frac{R}{L} \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 0 \\ 1/L \end{bmatrix} v(t)$$

$$i(t) = y(t) = \begin{bmatrix} 0 & 1 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + D v(t)$$

Matrices A, B, C, D give us the S-S representation.

how to choose state variables:

- ① must be a system variable (any signal that responds to input signal stimulus)
- ② must be linearly independent (each variable cannot be expressed as a linear combination of the other variables)

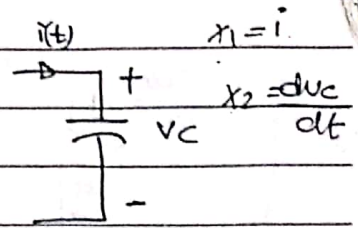
Super position.

Ex let us say you picked  $x_1, x_2, x_3$  as state variables,

If  $x_1 = 3x_2 + 2x_3$ . variables are not linearly independent.

III

Ex  $x_1 = 2x_2$  linearly dependent



$i(t) = C \frac{dv_C}{dt} \Rightarrow x_1(t) = C x_2(t)$   
linearly dependent.

(\*) Note: A set of objects  $x_i; (i=1, 2, \dots, m)$  are said to be linearly independent if:

$$\sum_{i=1}^m c_i x_i = 0 \quad \text{only if } c_i = 0 \text{ for all values of } i.$$

Note  $c_i \Rightarrow$  are constants  
( $i=1, 2, \dots, m$ )

obj 1) linear combination of variables and constants  
absolutely independent constants  
L. indep  $\Rightarrow$   $0 = \text{constants}$   $\Rightarrow$   $c_i = 0$

(5) A min # of state variables must be selected.

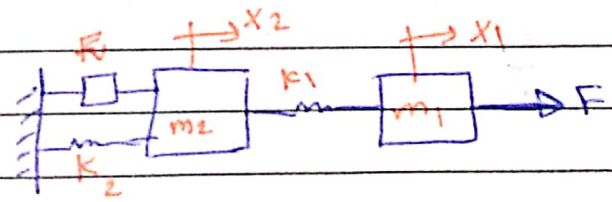
Rules: For a simple linear circuit:

# of state variables = # of energy storage elements (C, L)

For a translational mechanical system:

# of state variables = # of DOF  $\times 2$

# of DOFs = # of directions that independent masses move in.



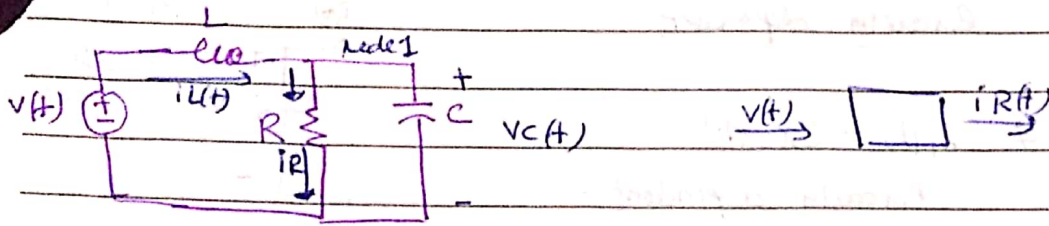
how to choose state variable for:

1) circuit: choose the differential variables of energy storage element.

2) Tran. mechanical system: choose state variables to be the position [displacement] and velocity of each point of linearly independent motion [DOF].

lec 63

(Ex 3.1 page 124)



⊗ state variables

Q: find SS representation:

$$\text{Let: } x_1 = v_C$$

$$x_2 = i_L$$

$$\text{Sol: } L \frac{di_L}{dt} = v_L = v - v_C \rightarrow \frac{di_L}{dt} = \frac{-v_C}{L} + \frac{v}{L}$$

$$C \frac{dv_C}{dt} = i_C$$

$$\dot{x}_2 = \frac{-x_1}{L} + \frac{v}{L} \quad \otimes$$

$$\dot{x}_1 = \frac{i_C}{C} = \frac{i_L - i_R}{C} = \frac{i_L}{C} - \frac{v_C}{RC}$$

$$\dot{x}_1 = \frac{-x_1}{RC} + \frac{x_2}{C} \quad \otimes$$

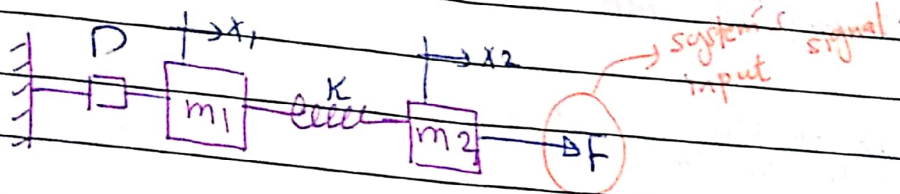
$$\therefore \begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} -\frac{1}{RC} & \frac{1}{C} \\ -\frac{1}{L} & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 0 \\ \frac{1}{L} \end{bmatrix} v(t)$$

$\begin{matrix} 2 \times 1 & & 2 \times 2 & & 2 \times 1 & & 2 \times 1 & & 2 \times 1 \end{matrix}$

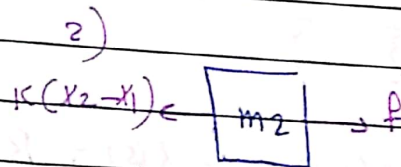
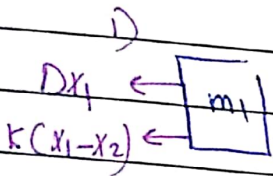
$$\therefore \begin{bmatrix} i_R \\ \end{bmatrix} = \begin{bmatrix} \frac{1}{R} & 0 \\ \end{bmatrix} \begin{bmatrix} x_1 = v_C \\ x_2 = i_L \end{bmatrix} + \begin{bmatrix} 0 \\ \end{bmatrix} v(t)$$

$\begin{matrix} 1 \times 1 & & 1 \times 2 & & 1 \times 1 & & 1 \times 1 \end{matrix}$

Ex 3.3 p: 128



Sol: draw a FRD:



We expect the state eq. to have 4 state variables.

$m_1$

$$\sum F_{x_1} = m_1 \ddot{x}_1 = -F(x_1 - x_2) - D\dot{x}_1$$

$m_2$

$$\sum F_x = m_2 \ddot{x}_2 = F - k(x_2 - x_1)$$

state vector:

$$= \begin{bmatrix} x_1 \\ \dot{x}_1 \\ x_2 \\ \dot{x}_2 \end{bmatrix} = \sum \begin{bmatrix} x_1 \\ v_1 \\ x_2 \\ v_2 \end{bmatrix}$$

$$\dot{\sum} = A \sum(t) + B F(t)$$

[Back to Book]

$$y = C \sum(t) + D F(t) = \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ v_1 \\ x_2 \\ v_2 \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \end{bmatrix} F(t)$$

$$\begin{bmatrix} \dot{x}_1 \\ \dot{v}_1 \\ \dot{x}_2 \\ \dot{v}_2 \end{bmatrix} = \begin{bmatrix} 0 & 1 & 0 & 0 \\ -k/m_1 & -D/m_1 & k/m_2 & 0 \\ 0 & 0 & 0 & 1 \\ k/m_2 & 0 & 0 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ v_1 \\ x_2 \\ v_2 \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ 0 \\ 1/m_2 \end{bmatrix} F(t)$$

$$\dot{x}_1 = v_1$$

$$\dot{x}_2 = v_2$$

### Sec 3.6 Finding T.F from s-s representation.

Given:  $\dot{x} = Ax + Bu \rightarrow (1)$   
 $n \times 1$     $n \times n$     $n \times 1$     $n \times m$     $m \times 1$

$y = Cx + Du \rightarrow (2)$   
 $p \times 1$     $p \times n$     $n \times 1$     $p \times m$     $m \times 1$

Taking  $\mathcal{L} \{ \}$

$$sX(s) - x(t=0) = AX(s) + BU(s)$$

$$(sI - A)X(s) = x_0 + BU(s)$$

$$X(s) = (sI - A)^{-1} x_0 + (sI - A)^{-1} BU(s)$$

Take:

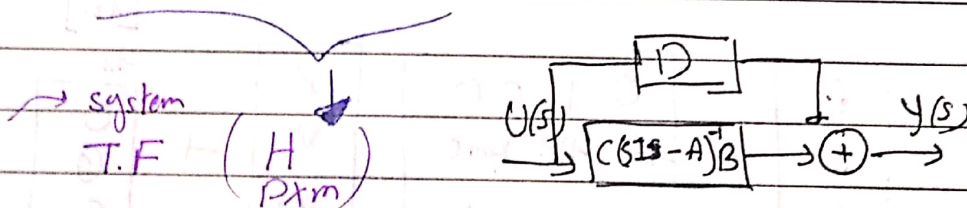
$$\mathcal{L} \{ \text{Eq (2)} \} \quad Y(s) = C X(s) + DU(s)$$

$$Y(s) = C [(sI - A)^{-1}] x_0 + C [(sI - A)^{-1}] BU(s) + DU(s)$$

$$= C(sI - A)^{-1} x_0 + [C(sI - A)^{-1} B + D] U(s)$$

⊗ to find T.F, let  $x(0) = 0$ :

$$Y(s) = [C(sI - A)^{-1} B + D] U(s)$$



① let  $D=0$  and let  $m=p=1$  (SISO)

$$Y(s) = C(sI - A)^{-1} B U(s)$$

$$\frac{Y(s)}{U(s)} = \frac{C \text{Adj}(sI-A) B}{\text{Det}(sI-A)}$$

polynomial in s

② MIMO system  
 $\frac{Y(s)}{U(s)}$  exists

②  $m \neq 1, P \neq 1$  (MIMO)

$$Y(s) = \left[ \begin{array}{c|c} C \text{adj} [sI-A] B & \\ \hline \det [sI-A] & \end{array} \right]_{p \times m} U(s)$$

only in zero is special case

$H_{p \times m}$  is called the system's **matrix transfer function**.

$$H = \begin{bmatrix} \frac{N_{11}(s)}{Q(s)} & \frac{N_{12}(s)}{Q(s)} & \dots & \frac{N_{1m}}{Q(s)} \\ \frac{N_{21}(s)}{Q(s)} & \frac{N_{22}(s)}{Q(s)} & \dots & \frac{N_{2m}}{Q(s)} \\ \dots & \dots & \dots & \dots \\ \frac{N_{p1}(s)}{Q(s)} & \dots & \dots & \frac{N_{pm}(s)}{Q(s)} \end{bmatrix}$$

by setting  $\det[sI-A] = 0$  we find poles.  
 $Q(s) = \det[sI-A]$

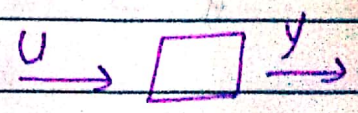
T.F.  $\frac{N}{Q}$   
 poles,  $\frac{N}{Q}$   
 zeros,  $\frac{N}{Q}$   
 zero

$$W_{m \times 1} = \begin{bmatrix} u_1 \\ u_2 \\ \vdots \\ u_m \end{bmatrix}$$

$$Y_{p \times 1} = \begin{bmatrix} y \\ \vdots \\ y_p \end{bmatrix}$$

lec 7 HW Ex 3.6 page 137

Sec 3.5 : Converting a DF to S.S representation, and system's implementation using integration.





Ex Given the DE  $\ddot{y}(t) + 2\dot{y}(t) + 3y(t) = 7u(t)$ .

Sol: Let:  $x_1 = y$

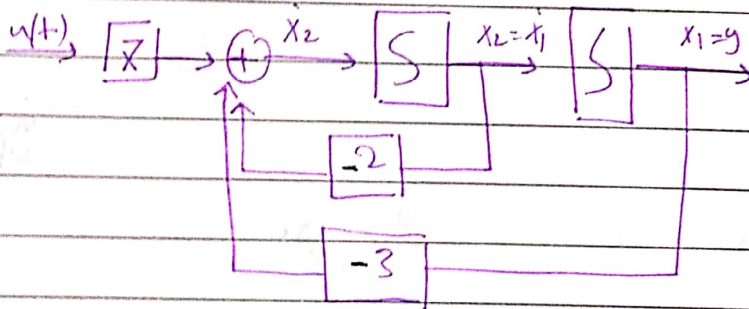
Let:  $x_2 = \dot{y} = \dot{x}_1 \rightarrow \boxed{\dot{x}_1 = x_2}$

$\dot{x}_2 = \ddot{y} = -3y - 2\dot{y} + 7u$  , let:  $x = \begin{bmatrix} x_1 \\ x_2 \end{bmatrix}$  ;  $\dot{x} = \begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix}$   
 $\dot{x}_2 = -3x_1 - 2x_2 + 7u$

$\dot{x} = \begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ -3 & -2 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 0 \\ 7 \end{bmatrix} u$

$y = x_1 = \begin{bmatrix} 1 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 0 \end{bmatrix} u$

In general, given DE:  $y^{(n)} + a_{n-1}y^{(n-1)} + \dots + a_1\dot{y} + a_0y = b_0u(t)$



Let:  $x = \begin{bmatrix} x_1 \\ x_2 \\ \vdots \\ x_{n-1} \\ x_n \end{bmatrix}$

S-S representation will be:

$\dot{x} = \begin{bmatrix} 0 & 1 & & \\ & 0 & 1 & \\ & & \ddots & \ddots \\ -a_0 & -a_1 & \dots & -a_{n-1} \end{bmatrix} x + \begin{bmatrix} 0 \\ 0 \\ \vdots \\ b_0 \end{bmatrix} u(t)$

controllable canonical form.

OR phase var form.

$y = \begin{bmatrix} 1 & 0 & 0 & \dots & 0 \end{bmatrix} x + \begin{bmatrix} 0 \end{bmatrix} u$

Ex Given  $\ddot{y} + 2\dot{y} + 3y = \dot{u} + u \dots$

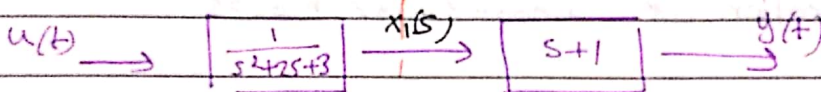


assuming zero I.C.s:  $\mathcal{L}\{\circledast\} =$

$$[s^2 + 2s + 3] Y(s) = [s + 1] U(s)$$

$$\frac{Y(s)}{U(s)} = \frac{s+1}{s^2 + 2s + 3} = \frac{1}{s^2 + 2s + 3} \cdot (s+1)$$

$$= \frac{X_1(s)}{U(s)} = \frac{Y(s)}{X(s)}$$



$$\frac{X(s)}{U(s)} = \frac{1}{s^2 + 2s + 3} \quad \mathcal{L}^{-1} : \begin{cases} \ddot{x}_1 + 2\dot{x}_1 + 3x_1 = u \\ \dot{x}_2 = -2x_2 - 3x_1 + u \end{cases}$$

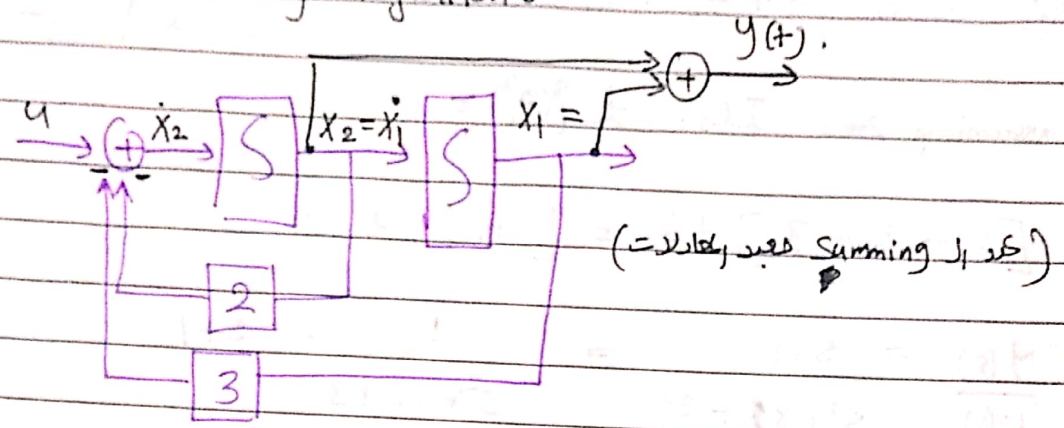
Def  $x_2 = \dot{x}_1 \Rightarrow$  state equation:  $x = \begin{bmatrix} x_1 \\ x_2 \end{bmatrix}$

$$\rightarrow \dot{x} = \begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ -3 & -2 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 0 \\ 1 \end{bmatrix} u \dots \text{s-eq} \checkmark$$

But  $Y(s) = [s+1] X_1(s) \xrightarrow{\mathcal{L}^{-1}} y(t) = \dot{x}_1(t) + x_1(t) = x_2 + x_1$

$$\therefore y(t) = \begin{bmatrix} 1 & 1 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} \dots \text{output-eq} \checkmark$$

Implementation using integration:



H.W Ex 3.4 & Ex 3.5 in T.B.

lec 8

linearization of a non-linear model:

⊗ linear S-S model :

$$\dot{x} = Ax + Bu$$

$$y = Cx + Du$$

Given the sys of equations:

$$\dot{x}(t) = f(x(t), u(t)) \neq Ax(t) + Bu(t)$$

$n \times 1$ 
 $n \times 1$ 
 $n \times 1$ 
 $m \times 1$

non-linear.

and  $y = h(x(t), u(t))$

non-linear.

[around a point of equi.]

∴ we obtain a linear S-S model by linearizing about an equilibrium point  $(x_0, u_0)$ , such that  $f(x_0, u_0) = 0$

let  $x(t) = x_0 + \Delta x(t)$  ;  $u(t) = u_0 + \Delta u(t)$

$$\dot{x}(t) \approx f(x_0, u_0) + \left. \frac{df}{dx} \right|_{x(t)=x_0} \Delta x(t) + \left. \frac{df}{du} \right|_{u(t)=u_0} \Delta u(t)$$

$n \times 1$        $n \times 1$        $n \times n$  matrix       $n \times 1$        $n \times m$  matrix       $n \times 1$

where  $\frac{df}{dx}$  is a matrix let:  $\frac{df}{dx} = \begin{bmatrix} \frac{\partial f_1}{\partial x_1} & \frac{\partial f_1}{\partial x_2} & \dots & \frac{\partial f_1}{\partial x_n} \\ \frac{\partial f_2}{\partial x_1} & \frac{\partial f_2}{\partial x_2} & \dots & \frac{\partial f_2}{\partial x_n} \\ \dots & \dots & \dots & \dots \\ \frac{\partial f_n}{\partial x_1} & \dots & \dots & \frac{\partial f_n}{\partial x_n} \end{bmatrix}$

$$f(x, u) = \begin{bmatrix} f_1(x, u) \\ f_2(x, u) \\ \vdots \\ f_n(x, u) \end{bmatrix} ; x(t) = \begin{bmatrix} x_1(t) \\ x_2(t) \\ \vdots \\ x_n(t) \end{bmatrix} ; x_0 = \begin{bmatrix} x_{01} \\ x_{02} \\ \vdots \\ x_{0n} \end{bmatrix}$$

$n \times 1$        $n \times 1$        $n \times 1$

$$u(t) = \begin{bmatrix} u_1(t) \\ \vdots \\ u_m(t) \end{bmatrix} ; \Delta x(t) = x(t) - x_0$$

$m \times 1$

$$\frac{df}{du} = \begin{bmatrix} \frac{\partial f_1}{\partial u_1} & \dots & \frac{\partial f_1}{\partial u_m} \\ \vdots & \dots & \vdots \\ \frac{\partial f_n}{\partial u_1} & \dots & \frac{\partial f_n}{\partial u_m} \end{bmatrix}, B = \left. \frac{df}{du} \right|_{\substack{x=x_0 \\ u=u_0}}, A = \left. \frac{df}{dx} \right|_{\substack{x=x_0 \\ u=u_0}}$$

$$\dot{x}(t) = f(x_0, u_0) + A \Delta x(t) + B \Delta u(t)$$

$$\dot{x}(t) = A \Delta x + B \Delta u ; \text{But } \dot{x}(t) = \Delta \dot{x}(t)$$

$$\Delta \dot{x}(t) = A \Delta x(t) + B \Delta u(t)$$

we can solve for  $\Delta x$

$$\Delta x(t) = x_0 + \Delta x(t)$$

Q What would you do to linearize a non-linear equation?

$$y(t) = h(x(t), u(t))$$

$$y(t) \approx h(x_0, u_0) + \left. \frac{\partial h}{\partial x} \right|_{\substack{x=x_0 \\ u=u_0}} \Delta x + \left. \frac{\partial h}{\partial u} \right|_{\substack{x=x_0 \\ u=u_0}} \Delta u$$

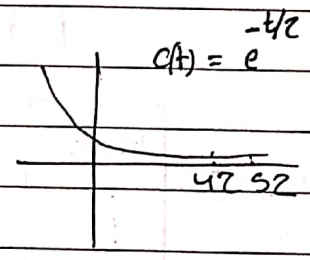
$$\Delta y = y(t) - h(x_0, u_0)$$

$$\Delta y = C \Delta x(t) + D \Delta u(t)$$

HW Ex 3.2 ✓  
page 139.

Chapter 4: 3

$$c(t) = e^{-st} = e^{-\frac{t}{15}}$$



what is time constant ?!

$e^{-t/\tau}$  time constant ; settling time =  $4\tau = \frac{4}{a}$

\* for 2nd order system:-

$$T(s) = \frac{K}{s^2 + 2\zeta\omega_n s + \omega_n^2} \Rightarrow T(s) = \frac{4}{s^2 + 2\zeta\omega_n s + \omega_n^2}$$

step response:  $c(t) = s-s \text{ value} + e^{-\zeta\omega_n t} \cos(\omega_d t + \phi)$

transient

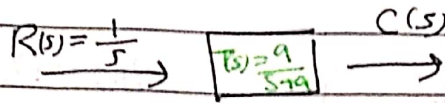
1st order: step response =  $1 - e^{-at}$

$$T_r = \frac{2.2}{a} = 2.2\tau$$

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$$T_s = 4\tau = \frac{4}{a}$$

Lec 9:



$$C(s) = T(s)R(s) = \frac{a}{s(s+a)}$$

$$C(t) = \mathcal{L}^{-1} \left\{ \frac{a}{s(s+a)} \right\} = \mathcal{L}^{-1} \left\{ \frac{1}{s} + \frac{-1}{s+a} \right\} = 1 - e^{-at}$$

$= 1 - e^{-t/\tau} \quad ; \tau = \frac{1}{a}$

after  $4\tau$ ,  $C(t)$  will reach steady state =  $C_p(t) + C_n(t)$

Forced response  
natural response  
(transient)

∴ after for  $t > 4\tau$ ,  $C(t) \approx C_p(t)$

⊛ For the 1st order system, steady-state step response equals 1.

DC gain

$$\lim_{t \rightarrow \infty} C(t) = \lim_{s \rightarrow 0} s C(s) = \lim_{s \rightarrow 0} s R(s) T(s) = \lim_{s \rightarrow 0} s \frac{1}{s} T(s)$$

$$\hookrightarrow \lim_{s \rightarrow 0} T(s) = \lim_{s \rightarrow 0} \frac{a}{s+a} = \frac{a}{a} = 1$$

⊛ Steady state response  $\equiv$  DC-gain of system. #

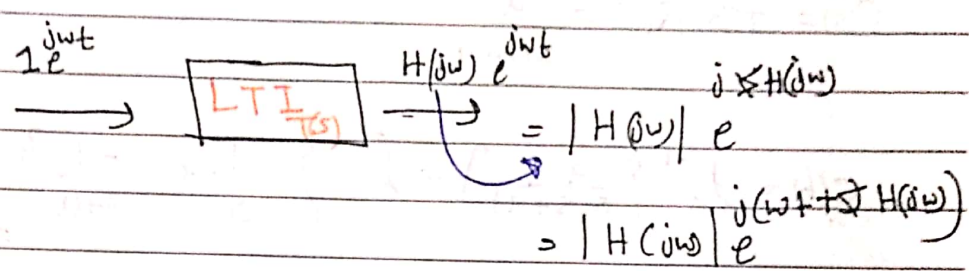
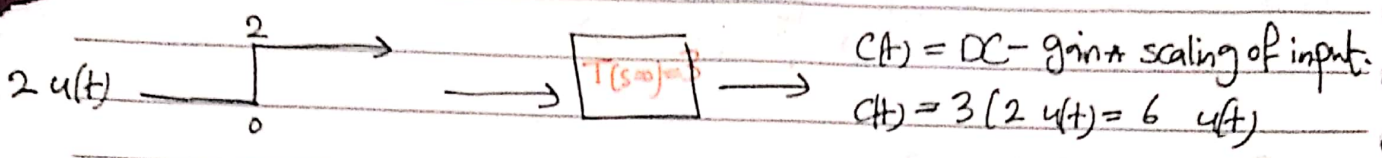
Recall Freq Response =  $T(s) \Big|_{s \rightarrow j\omega} = T(j\omega)$

$|T(j\omega)|$  = system gain for an input frequency  $\omega = \omega_0$ .

$$\text{DC-Gain} = |T(j\omega = 0)| = |T(s=0)| \quad \therefore \text{DC gain} \quad \otimes$$

TF  $\Big|_{\omega=0}$

DC gain  $\Big|_{\omega=0}$



2nd order underdamped systems:  $0 < \zeta < 1$

Block diagram: Input  $R(s) = \frac{1}{s}$  enters a block. Output  $C(s) = R(s) T(s) = \frac{\omega_n^2}{s(s^2 + 2\zeta\omega_n s + \omega_n^2)} = \frac{\omega_n^2}{s(s+p)(s+p^*)}$ .

Partial fraction expansion:  $= \frac{K_1}{s} + \frac{K_2}{s-p} + \frac{K_3}{s+p^*}$

$p = -\zeta\omega_n + j\omega_d$   
 $= -\zeta\omega_n + j\omega_n \sqrt{1-\zeta^2}$

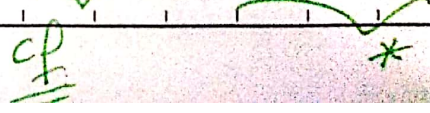
$s^2 + 2\zeta\omega_n s + \omega_n^2 = (s-p)(s-p^*)$   
 $\omega_d = \omega_n \sqrt{1-\zeta^2}$

$(s-1)(s-2) = s^2 - 3s + 2$  roots  $s = (1, 2)$

$\mathcal{L}^{-1} \left\{ \frac{K_1}{s} + \frac{K_2}{s-p} + \frac{K_3}{s-p^*} \right\}$   
 $= (K_1 + K_2 e^{pt} + K_3 e^{p^*t}) u(t)$

(\*) Try an example, to find that  $K_3 = K_2^*$

step response  $C(t) = K_1 u(t) + [K_2 e^{pt} + K_3 e^{p^*t}]$



has a phase shift

\*  $[k_2 e^{p_1 t} + k_3 e^{p_2 t}]$ :  $C_t$  (transient response) let  $k_2 = Ae$

$$= [A e^{-\zeta \omega_n t} e^{j \omega_d t} + A e^{-\zeta \omega_n t} e^{-j \omega_d t}]$$

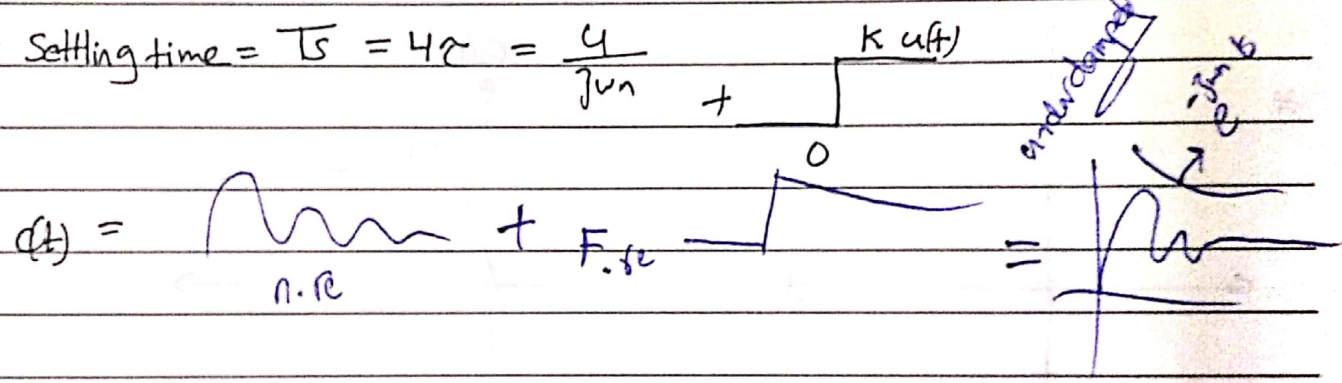
$$= A e^{-\zeta \omega_n t} [e^{j(\omega_d t + \theta)} + e^{-j(\omega_d t + \theta)}]$$

$$= A e^{-\zeta \omega_n t} 2 \cos(\omega_d t + \theta)$$

$$= A e^{-t/\tau} 2 \cos(\omega_d t + \theta) \quad \text{where } \tau = \frac{1}{\zeta \omega_n}$$

page 171

Setting time =  $T_s = 4\tau = \frac{4}{\zeta \omega_n}$



Lec 10: (HW) Ex 4.5 page 172.

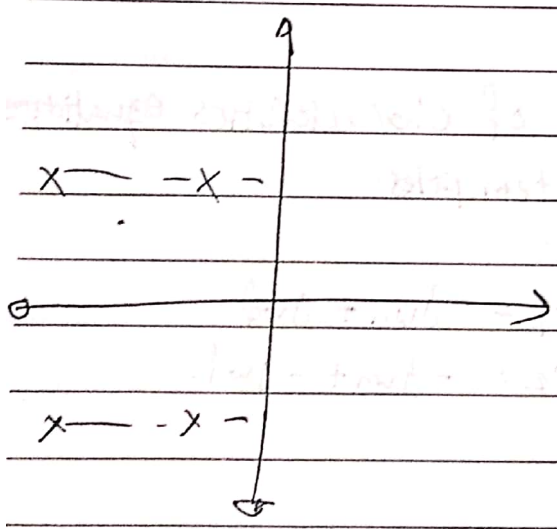
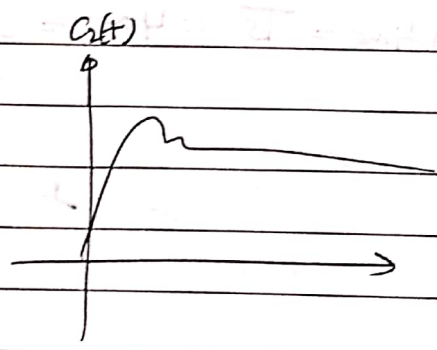
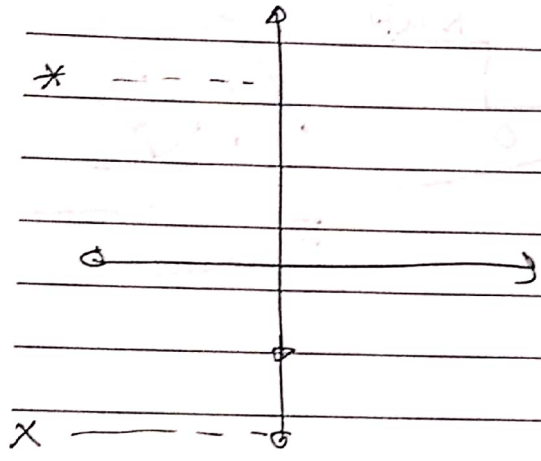
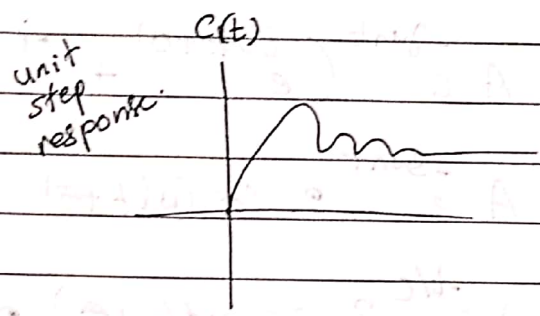
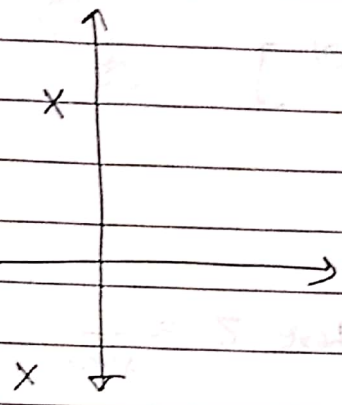
$$T(s) = \frac{k \omega_n^2}{s^2 + 2\zeta \omega_n s + \omega_n^2} \quad \Rightarrow \text{roots of characteristics equation = system poles.}$$

[underdamped system]  $0 < \zeta < 1$  ,  $p_1 = -\zeta \omega_n + j \omega_d$   
 $p_2 = -\zeta \omega_n - j \omega_d$

(HW) Ex 4.6 page 180.



See 4.7: Approximating the transient response of a system having more than 2 poles.

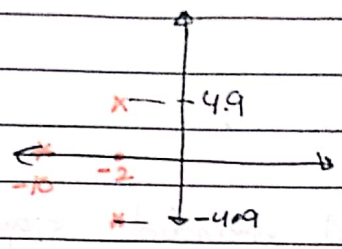


$$T(s) = \frac{K}{(s^2 + 2\zeta\omega_n s + \omega_n^2)(s^2 + 2\zeta_2\omega_{n2}s + \omega_{n2}^2)}$$

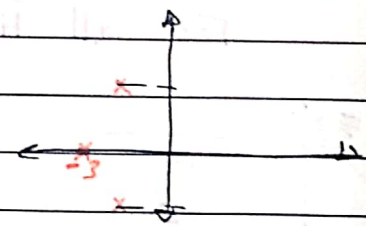
And step response:  $c(t)$ :

$$c(s) = \frac{K_1}{s} + \left[ \frac{K_2}{s} + \frac{F_3}{s} \right]$$

$$T_1(s) = \frac{24.542}{s^2 + 4s + 24.542}$$



$$T_2(s) = \frac{245.42}{(s+10)(s^2 + 4s + 24.542)}$$



$$T_3(s) = \frac{23.626}{(s+3)(s^2 + 4s + 24.542)}$$

→ And 2nd order approximation for  $T_2$  and  $T_3$  :

$$T(s) \approx \frac{24.542}{s^2 + 4s + 24.542}$$

~~A~~  $T_2 \approx T_1$  approx A

$T_3 \approx T_1$  approx B  $\therefore$  A is Better than B.

Lee 11!

check Lee 11 PDF.

Second:

Lec 12:

## Ch 6:

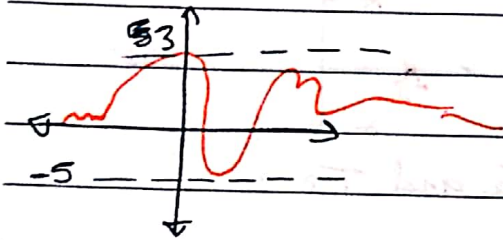
Definitions:

→ +ve real number.

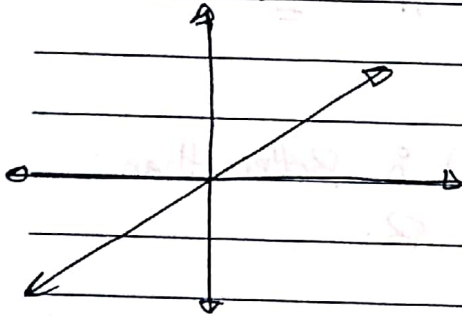
① A bounded signal,  $r(t)$ , is such that  $|r(t)| < B$   
For all time,  $t$ .

→ for all time

$$|r(t)| < 5$$



$$r_2(t) = t$$



we can't find a +ve real number,  $B$ , that satisfies  $|r_2(t)| < B$ .

$\therefore r_2(t)$  is unbounded.

as  $t$  goes to  $\infty$ ,  $r_2(t)$  goes to  $\infty$

inp. sig. → LTI → out. sig.

\* Stability definitions:

① a system is stable if for any bounded input, the output signal will be bounded. **[BIBO]** = That is system is BIBO stable.

② A system is marginally stable, if its output is bounded for some choice of input signal, but output is unbounded for other choices of input signals.

③ A system is unstable, if the output is unbounded for any choice of bounded input.

\* Stability For LTI systems:

$r(t) \rightarrow [LTI] \rightarrow c(t)$

Recall that:

$c(t) = c_{ft}(t) + c_n(t)$   
Force      transient

A LTI system is stable if transient response  $(c_n(t))$  goes  $\rightarrow 0$  as  $t \rightarrow \infty$

Ex  $T(s) = \frac{1}{s^2 + 2\zeta\omega_n s + \omega_n^2}$  ; poles of system

for underdamped case:  $(s^2 + p)(s + p^*)$   $s_{1,2} = -\zeta\omega_n \pm \omega_n \sqrt{\zeta^2 - 1}$  (general form)  
 $\rightarrow$  for underdamped systems:

$s_{1,2} = -\zeta\omega_n \pm j\sqrt{1 - \zeta^2}$

Ex :  $T(s) = \frac{1}{s^2 + s + 1}$

stable :  $e^{-\frac{1}{2}t}$  : expon. decaying.

$Re\{1,2\} = -\frac{1}{2}$

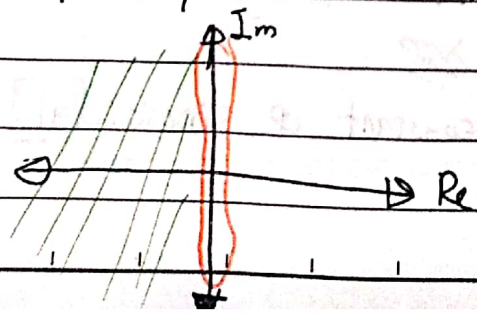
Ex  $T_2(s) = \frac{1}{s^2 - s + 1}$

unstable :  $e^{\frac{1}{2}t}$  : expon. growing.

$Re\{1,2\} = \frac{1}{2}$

A LTI sys is stable if all sys poles have negative real part. That is, all poles are in LHP.

the open.



⊗ poles have real part less than 0 are stable.  
 $< 0$

⊗ Real part of the pole <sup>in a stable system</sup> tells us about the exponential decay of (transient response).

⊗ A LTI system is unstable iff <sup>transient response</sup>  $C_n(t) \rightarrow \infty$  as  $t \rightarrow \infty$  (exponentially grows)

⊗ A LTI sys is marginally stable iff  $C_n(t)$  (transient) neither grows, nor decays with time, rather transient remains constant or oscillates as  $t \rightarrow \infty$

Note Pole multiplicity:

$$TF \equiv T(s) = \frac{1}{(s+p_1)^{z_1} (s+p_2)^{z_2}}$$

$z_2$ : is the multiplicity of  $p_2$ .

→ Unstable systems: have poles with ±ve real part <sup>expo growing / RHP</sup>

and/or imaginary-axis poles with multiplicity  $z > 1$ .

→ marginally stable systems: <sup>MS</sup> have <sup>at least</sup> poles on the  $j\omega$  axis with multiplicity of one, and/or poles with -ve real part <sup>expo. decaying [LHP]</sup>.

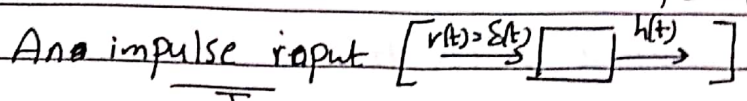
Note Imaginary-axis poles with multiplicity  $z$  ( $z=1, 2, \dots$ ) produce a transient response of the form:

$$C_n(t) = \sum_{k=1}^z A_k t^{k-1} \cos(\omega_k t + \phi_k)$$

IF  $z=1$ , then system is M.S. - [constant or sinusoidal]

impulse response.

Notes



is a special-case theoretical input.

$\delta(t)$  is unbounded  $\rightarrow \infty$  but absolutely integrable.

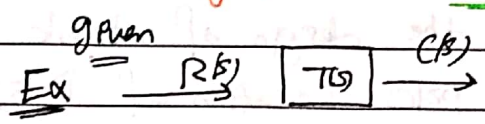


$$\int_{-\infty}^{\infty} |\delta(t)| dt = 1$$

Impulse response of LTI systems explain the transient behaviour of the system.

LTI system is stable if  $\int_{-\infty}^{\infty} |h(t)| dt < \infty$  (Integrable).

$\hookrightarrow$  is absolutely integrable.



$$T(s) = \frac{1}{s^2(s+2)} \rightarrow \text{Find impulse response.}$$

Sol let  $R(s) = 1$  ( $r(t) = \delta(t)$ )

$$\text{Impulse response} = R(s) T(s) = T(s) = \frac{1}{s^2(s+2)}$$

$$= \frac{0.25}{s+2} + \frac{-0.25}{s} + \frac{0.5}{s^2}$$

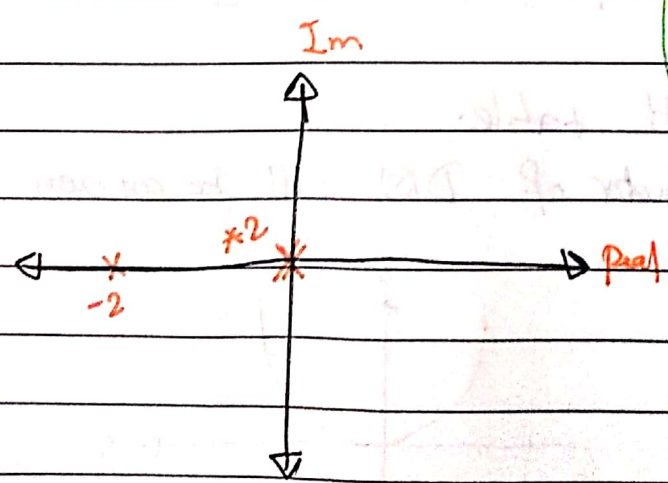
transient due to pole at  $\underline{-2}$

transient due to double (2=2) poles at origin

$$\mathcal{L}^{-1} \left[ \frac{1}{s^2(s+2)} \right] = (-0.25 + 0.5t)u(t).$$

$\hookrightarrow$  so: as  $t \rightarrow \infty$  transient will  $\rightarrow \infty$

$\therefore$  **unstable system.**



unstable due to (2=2) for pole: 0 at origin without any pole at the

RHP

Lec 13:

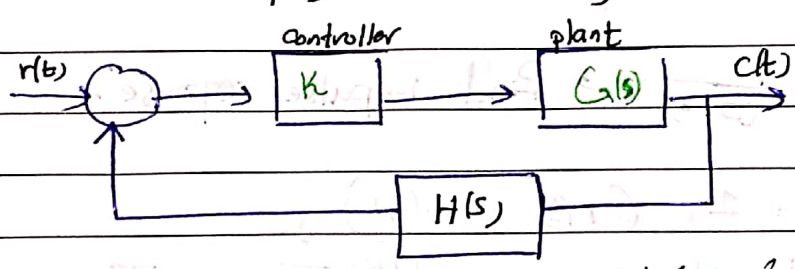
Routh-Hurwitz method : (R-H)

Determines stability of a system by analyzing the system's characteristic polynomial  $D(s)$ . T.F

See table 6.2 page 304

R-H method finds No. of poles in open-LHP [Complex-plane], open RHP, and No of poles on jw-axis.

The strength of R-H method is in the design of stable systems, rather than simply determining poles location of  $D(s)$ .



See ex 6.1 304

$r(t) \rightarrow \boxed{T(s)} \rightarrow c(t) \therefore T(s) = \frac{N(s)}{D(s)K}$

از این بیرون سوار پoles

از این بیرون سوار پoles

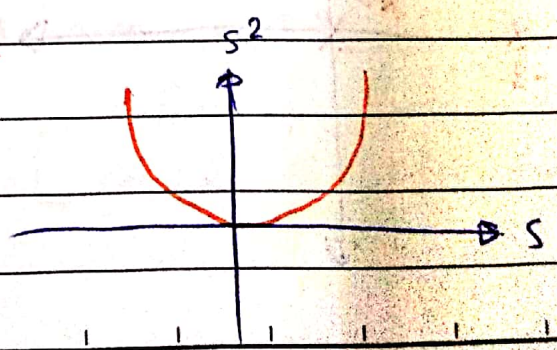
Special cases of Routh - tables:

1] zero in first column of R-H table.

Ex 6.2 & Ex 6.3 page [306-307].

2] Entire row of zeros in R-H table.

means that we have that a factor of  $D(s)$  will be an even polynomial.



Case even  $\rightarrow$  analyzing row of zeros  $\otimes$   
polynomial

$\rightarrow$   $\omega$ -axis, as poles,  $\rightarrow$   $\omega$  axis sign change  $\otimes$

Ex1  $s^2$  is <sup>even</sup> polynomial  $\rightarrow$  even polynomials include even powers of  $s$ .

Ex2  $s^4 + 3s^2 + 2$ , is even in  $s$ .

$\rightarrow$  Roots of even polynomials have symmetry about the origin of  $s$ -plane. [see Figure 6.5 P: 309]  $\checkmark$

$\rightarrow$  To solve Problem in R-H table, previous row differentiation.

Ex 6.5  $\checkmark$   $\otimes$  No row of zeros in R-H table  $\rightarrow$  No roots on  $j\omega$ -axis (poles)

$\otimes$  Row of zeros in R-H table  $\leftarrow$  poles on  $j\omega$ -axis.

$\rightarrow$   $\omega$ -axis, as poles,  $\rightarrow$   $\omega$  axis sign change  $\otimes$

$\rightarrow$  Design Example Ex 6.9 (page 316):

Q) Given  $S$ - $S$  model, how would you build R-H table?

① we find the characteristic polynomial  $D(s) = \det(sI - A)$

EMI of # Ch 6 T.F.  $\rightarrow$  if  $\checkmark$

Dania cheek

SA 6.2

P: 311.

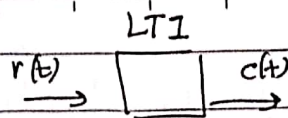


$E(s)$ : -ve real part  $\odot$   
 original pole  $\odot$   
 $E(s) \leftarrow$   $\infty$   $\rightarrow$   $\infty$

Lec 14:

Ch # 2:

Finite



Definition: error =  $e(t) = r(t) - c(t)$

define  $ess = \lim_{t \rightarrow \infty} e(t)$  [steady state error]

It's desirable that  $ess = 0 \iff \lim_{t \rightarrow \infty} c(t) = r(t)$

Final value theorem: (FVT) :-

Let:  $E(s) = R(s) - C(s) \iff e(t) = r(t) - c(t)$

$$ess = \lim_{t \rightarrow \infty} e(t) = \lim_{s \rightarrow 0} s E(s)$$

RHP  $\rightarrow$  poles  $\odot$

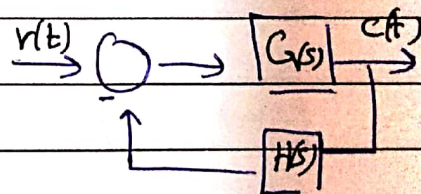
not valid

theory  $\rightarrow$  [with imaginary part]  $\rightarrow$  poles  $\odot$

FVT: is valid only if  $E(s)$  has poles with -ve real part, and/or poles on the origin of the s-plane. If  $E(s)$  has more than one poles at origin, then  $ess$  will be infinite.

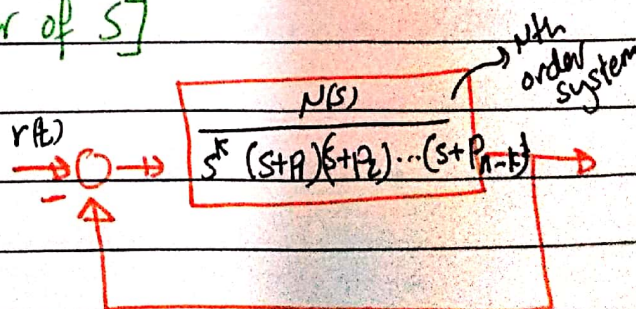
Recall:

$$T(s) = \frac{C(s)}{R(s)} = \frac{G(s)}{1 + G(s)H(s)}$$

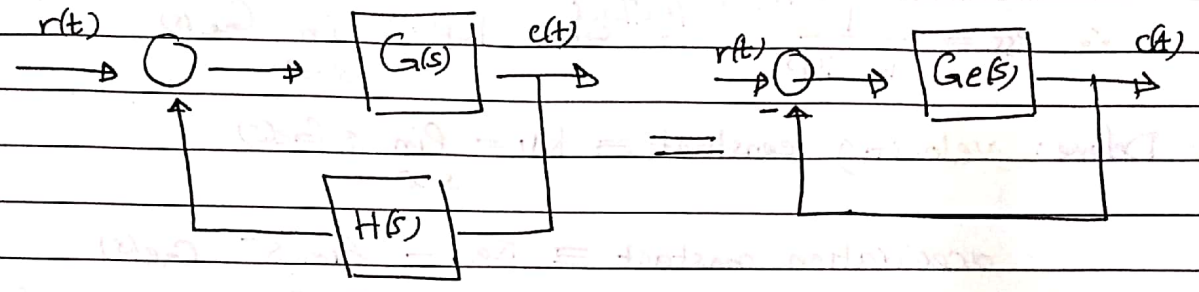


System type: is Number of Pure integrators in the forward path of a unity-feedback system. [power of s]

K is the system type.



\* (Q) How to determine system type with a non-unity feedback.



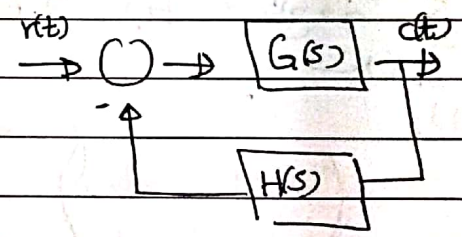
$$T(s) = \frac{G(s)}{1 + G(s)H(s)} \implies T(s) = \frac{G_c}{1 + G_c} = \frac{G_c}{1 + G_c H}$$

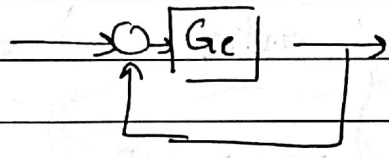
$\therefore G_c = \frac{G}{1 + G_c H - G}$  ~~✗~~

Ex 7.8 (p354) system type, and ess for a unity step input.

↳ Sol: to find system type,

find  $G_c = \frac{G}{1 + G_c H - G}$



then  $\rightarrow$   Given  $R(s) = \frac{1}{s}$  (unity-step input)

$$\begin{aligned} \text{ess} &= \lim_{t \rightarrow \infty} e(t) = \lim_{s \rightarrow 0} s E(s) \\ &= \lim_{s \rightarrow 0} s [R(s) - T(s)] \\ &= \lim_{s \rightarrow 0} s [R(s) - R(s) T(s)] \\ &= \lim_{s \rightarrow 0} s R(s) [1 - T(s)] \end{aligned}$$

But  $R(s) = \frac{1}{s}$  :  
 $\text{ess} = \lim_{s \rightarrow 0} [1 - T(s)]$

sys stable

$\int_0^{\infty} e^{-\sigma t} dt$  finite  $\lim_{s \rightarrow 0} G_e(s)$

9

$$L = \lim_{s \rightarrow 0} \left[ 1 - \frac{G_e}{1+G_e} \right] = \lim_{s \rightarrow 0} \frac{1}{1 + \lim_{s \rightarrow 0} G_e}$$

unit step input / unity feedback  $\rightarrow$  position constant (static error constant)

$$\therefore e_{ss} = \frac{1}{1 + K_p} \quad \text{where } K_p = \lim_{s \rightarrow 0} G_e(s)$$

\* Define: velocity constant  $\equiv K_v = \lim_{s \rightarrow 0} s G_e(s)$

acceleration constant  $\equiv K_a = \lim_{s \rightarrow 0} s^2 G_e(s)$

\* Table 7.2 (page 348) Table assume a unity-feedback

HW Skill Assessment Exercise 2.2 P (348)

Ex 2.5 P 349 ✓

Range of  $K$  for stability  $\rightarrow$  Ex 2.6 P 350 ✓ + 2.3 ✓

Range of  $K$  for stability

END of ch #2

Input	steady-state error formula	Type 0 static error constant	Type 1 s-e constant	Type 2 s-e constant
step, $u(t)$	$\frac{1}{1+K_p}$	$K_p = \text{constant}$	$\frac{1}{1+K_p}$	$K_p = \infty$
$\frac{1}{s^2} \rightarrow$ Ramp, $t u(t)$	$\frac{1}{K_v}$	$K_v = 0$	$\infty$	$K_v = \text{constant}$
Parabola, $\frac{t^2}{2} u(t)$	$\frac{1}{K_a}$	$K_a = 0$	$\infty$	$K_a = \text{constant}$

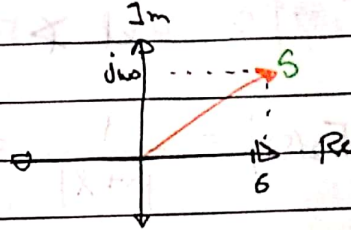
static gain variable  $0 < K < \infty$  poles in s-plane  $[closed\ loop\ poles]$

lec 152

Ch # 8: Root locus:

Intro: vector representation of complex number.

Given  $s = \sigma + j\omega$



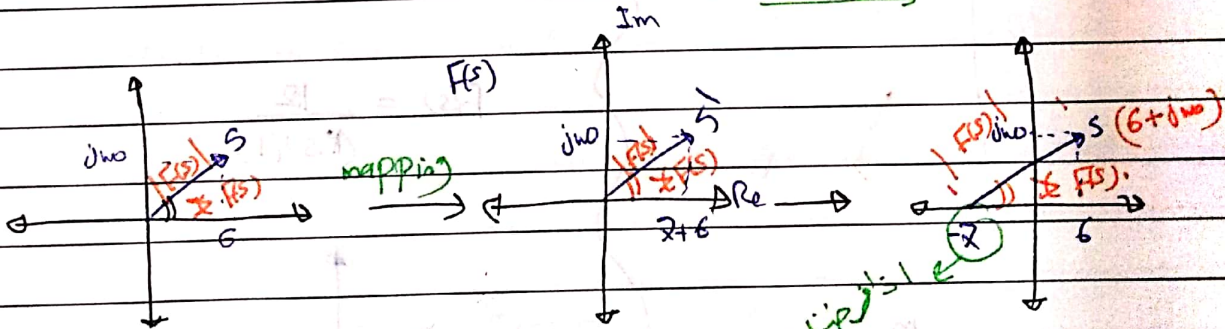
Mapping a value of  $s$  through a function  $F(s)$ .  
 OR  
 [function]  $F(s)$

Representation of  $|F(s)|$  (magnitude / gain) and  $\angle F(s)$  (phase shift), That is:  
 $F(s) = |F(s)| e^{j\angle F(s)}$  (gain + phase)

Ex Given  $F(s) = s + 2$ , where  $s = \sigma + j\omega$

$z = F(s) = (\sigma + 2) + j\omega$   
 $s = \sigma + j\omega$

translation with 2 units



mag + phase

function  $F(s)$  part

TF:  $(\sigma + j\omega)$

TF = TF poles

Ex 385 + S.A 8.1 : two ways to solve:

① Directly substituting the point in  $F(s)$ .

② Calculate the result using vectors.

Ex 2  $F_2(s) = \frac{1}{s+x} = \frac{1}{F(s)}$

$|F_2(s)|$  ?? and  $\angle F_2(s)$  given  $s = 6 + j\omega$

$|F_2(s)| = \frac{1}{|s+x|} = \frac{1}{|F(s)|} \Rightarrow \angle F_2(s) = -\angle F(s)$

In general IF  $F(s) = \frac{\prod_{i=1}^m (s+z_i)}{\prod_{j=1}^n (s+p_j)}$  then:

زیر صفر (zeros) → Poles  
صفر (zeros) → Poles

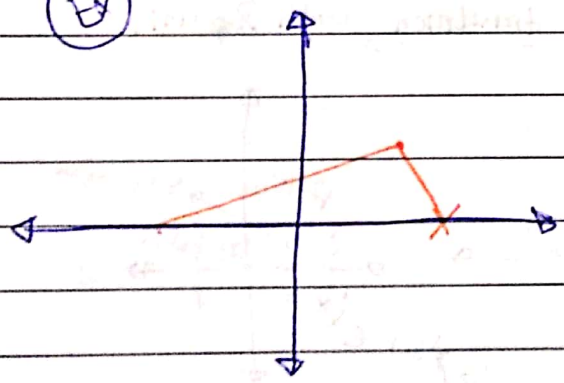
$|F(s)| = \frac{\prod_{i=1}^m |s+z_i|}{\prod_{j=1}^n |s+p_j|}$

length of  $\vec{v}_i$  drawn from 0 at  $-z_i$  to the point  $s$

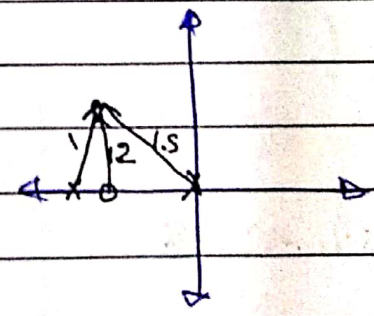
length of  $\vec{v}_j$  drawn from pole at  $-p_j$  to the point  $s$

صفر (zeros) → Poles  
صفر (zeros) → Poles  
magnitude

Ex



$F(s) = \frac{12}{(s)(1)}$

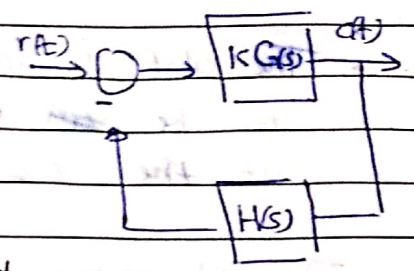


Also  $\angle F(s) = \sum_{i=1}^m \angle (s+z_i) - \sum_{j=1}^n \angle (s+p_j)$

Ex 8.1 page 385 ✓

F.W.  $0 < K < \infty$   
 Path

Given C.L.  $T(s) = \frac{K G(s)}{1 + K (G(s)H(s))}$   
 open loop transfer function O.L. TF  
 O.L. system



Root Locus: is the path each C.L. pole will traverse as the value of gain  $K$  is increased.

Let  $G(s) = \frac{N_G(s)}{D_G(s)}$  ,  $H(s) = \frac{N_H(s)}{D_H(s)}$

TF zeros: zeros of  $N_G(s)$  , poles of  $D_H(s)$

$\therefore T(s) = \frac{K G(s)}{1 + K G(s)H(s)} = \frac{K N_G(s) D_H(s)}{D_G(s) D_H(s) + K N_G(s) N_H(s)}$

as  $K \rightarrow 0$  ,  $T(s) \rightarrow \frac{\epsilon \epsilon}{D_G(s) D_H(s) + \epsilon \times 0}$  (TF pole)  
 $\hookrightarrow$  very small.

poles of C.L. system as  $(K \rightarrow 0) =$  roots of  $D_G(s) D_H(s)$   
 $=$  poles of  $G(s)H(s)$   
 open loop poles  $\odot =$  poles of O.L. systems.

as  $K \rightarrow \infty$  ,  $T(s) = \frac{K N_G(s) D_H(s)}{\epsilon + K N_G(s) N_H(s)}$

poles of C.L. system as  $K \rightarrow \infty =$  roots of  $N_G(s) N_H(s)$   
 $=$  zeros of O.L. system  $G(s)H(s)$   
 $\odot$

Lec 16:-

محل الأقطاب (Poles) في السطوح (S) في دالة النقل (TF) هي قيم  $s$  التي تجعل المقام يساوي صفرًا.

The plot of root locus starts (when  $K \rightarrow 0$ ) at the ~~any~~ O.L poles, and ends as ( $K \rightarrow \infty$ ) at the O.L zeros.

⊗ If O.L system (TF)  $G(s)H(s) = \frac{1}{s}$ , then:  
O.L system has:

- ① 1 finite pole at  $s=0$
- ② 1 infinite zero at  $s=\infty$  [  $\lim_{s \rightarrow \infty} \frac{1}{s} = 0$  same effect on O.L TF as finite zeros in O.L TF. ]

Ex ⊗ If O.L TF  $G(s)H(s) = \frac{s+1}{(s+2)(s+3)(s+7)}$

Then:  $G(s)H(s)$  has:

- ① 3 finite poles at  $-2, -3, -7$ .
- ② 1 finite zero at  $-1$ , and 2 zeros at  $s=\infty$ .

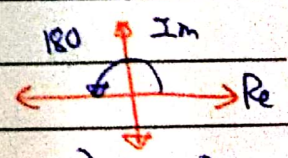
⊗ Points in S-plane that lie on root Locus?

C.L TF =  $\frac{K G(s)}{1 + K G(s)H(s)}$

poles of C.L system = roots of  $[1 + K G(s)H(s)]$   
= values of  $s$  that satisfy  $1 + K G(s)H(s) = 0$

$K G(s)H(s) = -1$

$\leftrightarrow |K G(s)H(s)| = 1$  and  $\angle (K G(s)H(s)) = (\pm 1) 180^\circ$   
 $K = 0, \pm 1, \pm 2, \dots$   
 $= \angle G(s)H(s)$



is  $K$  positive constant

SA: 8.2 ✓

⊗ Read 8.5 and study picture 8.13 page 396.

14

Ex 8.4

Points

⊗ Points that lie on root locus are values of  $s$  that satisfy:

$$\angle G(s)H(s) = (2k+1)180^\circ; k=0, \pm 1, \pm 2, \dots$$

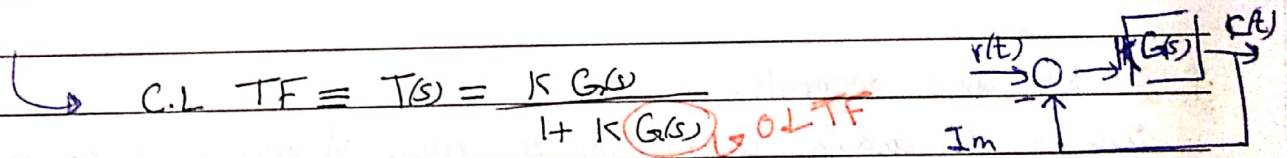
where value of gain  $k$  at each point root locus is:

$$k = \frac{1}{|G(s)H(s)|} = \frac{1}{|F(s)|} = \frac{\prod_{j=1}^n |u_j - s|}{\prod_{i=1}^m |v_i - s|}$$

⊗ HW Review [study] see 8.4, and Sec 8.5 → R.L. list 1-5 pages 391-392

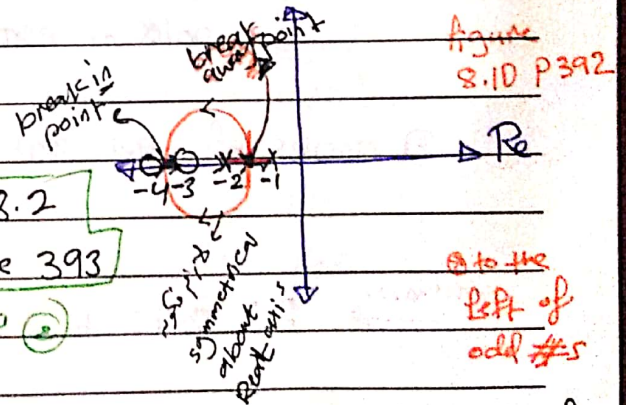
lec 12:

Ex let  $G(s) = \frac{(s+3)(s+4)}{(s+1)(s+2)}$ ,  $H(s) = 1$   
page 392



C.L TF  $\equiv T(s) = \frac{K G(s)}{1 + K G(s)}$

⊗ sketching the root locus:

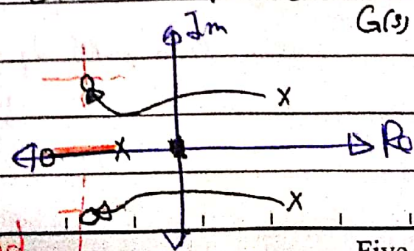
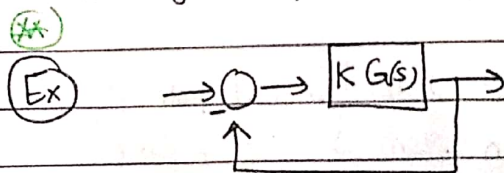


⊗ Behaviour at Infinity: Solve Ex 8.2

Eq =  $\frac{1}{s} \frac{1}{s} \dots$ ; real axis intercept:  $\frac{1}{s} \frac{1}{s} \dots$  Page 393

⊗ Angles of arrival / departure

o.l. zeros and poles (zeros and poles of  $G(s)$ )



R.L. to the left of  $P$  if  $P$  is odd # of real axis pole and zero.



Sec 9.4:

Sketching the Root Locus:-

Rules to draw R-L sketch with minimal calculations:

[عناصير = poles و صفرات = zeros و كسب = gain Rule) و كذا]

1) Number of branches: \* each C.L pole moves as the gain K varies \*

\* is the path of that the pole traverse.

Branches

WH

Rule: # of poles = # of CL poles of R-L sketch

2) Symmetry:

Rule: the R.L is symmetrical about the Real axis

3) Real axis segments:

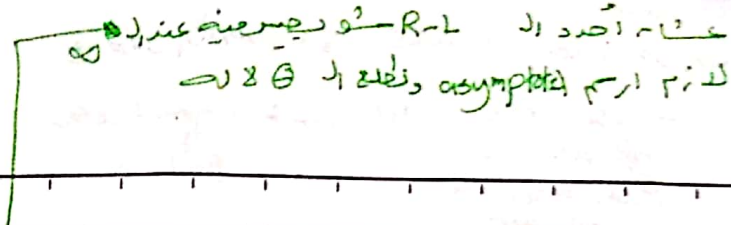
Rules: 1) angular contribution of a pair of open loop complex poles or zeros is zero. [سواء كانا صفرين او قطبين]

2) a) angles of real-axis pole or zeros is [0 or 180] Right Left

conclusion -> R-L exists to the left of an odd pole or zero Real-axis finite

4) Starting and ending points:

o.l poles  $\downarrow$  as  $K > 0$  ~~to~~  $\downarrow$  C.L pole  $\downarrow$   $K > \infty$   $\downarrow$  o.l zero  $\downarrow$   $\downarrow$   $\downarrow$



(5) Behavior at Infinity: (X) IF the function approaches  $\infty$  as  $s \rightarrow \infty$  then It has a pole at  $\infty$ .

(\*) IF the function approaches 0 as  $s \rightarrow \infty$  then It has a zero at  $\infty$ .

if the open loop poles are more than open loop zeros then there must be zeros at  $\infty$ .

Steps when sketching R-L: go to example 8.2 and S.A 5.3

(1) Calculate the asymptotes using:  $\sigma_a = \frac{\sum \text{finite poles} - \sum \text{finite zeros}}{\# \text{ finite poles} - \# \text{ finite zeros}}$

(2) Calculate the angles of the lines of intersect at  $\sigma_a$  using:

$$\theta_a = \frac{(2k+1)\pi}{\# \text{ finite poles} - \# \text{ finite zeros}}$$

For different values of  $k: 0, 1, 2, \dots$

ف عدد = عدد =

Note # of lines =  $\Delta [\text{finite poles} - \text{finite zeros}]$ .

(\*) to have a better picture of R-L: We can find angles of arrival and angles of departure.

~~real axis~~ ~~real axis~~

and calculate

(2) jW-axis crossings

stability

(3) real-axis break away and break in points

splane دے سکتے ہیں اور Real axis پر segment, اور

to the left of an odd number of segments of Real axis pole or zero

Lec 18:

18

Ex 8.7 R-L using matlab page 405:-

note

O.L TF =  $\frac{K(s+15)}{s(s+1)(s+10)}$   
 $= \frac{K(s+15)}{s^3+11s^2+10s}$

Ex 8.1  
 $sys = n(s)/d(s)$   
 rlocus(sys) will plot values of s that satisfy  $d(s) + K n(s) = 0$   
 $0 < K < \infty$

write using matlab:-

$numgh = [1 \ -4 \ 20]$

$dengh = [1 \ 6 \ 8]$

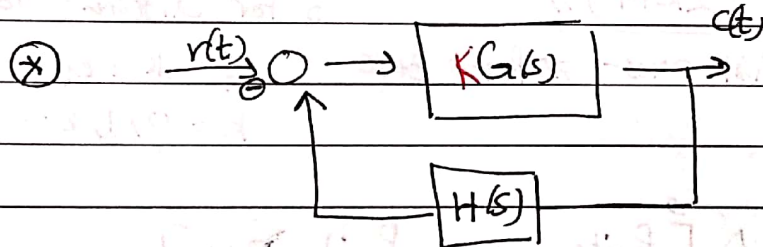
$dengh = poly([-2 \ -4])$

should give same result

TF  $\Rightarrow G = tf(numgh, dengh) \rightarrow$  gain کے جواب میں

$\Rightarrow rlocus(G) \rightarrow$  R-L کے function

$\Rightarrow doc rlocus \rightarrow$  command کے



C.L TF =  $\frac{KG(s)}{1 + KG(s)H(s)}$   
 O.L TF

From Lec 15:

Let  $G(s) = \frac{N_G(s)}{D_G(s)}$ ,  $H(s) = \frac{N_H(s)}{D_H(s)}$

substitute (\*) in eq (1)

$\rightarrow TF = \frac{K N_G(s) D_H(s)}{D_G(s) D_H(s) + K N_G N_H}$ , O.L TF =  $sys = \frac{N_G N_H}{D_G D_H + K n(s)}$

C.L TF =  $\frac{K N_G D_H}{d(s) + K n(s)}$

Back to Ch # 4 page 176: OS/ کے

$\cos \theta = \dots$

Back to 179

Five Apple

$\text{sgrid}(z, \omega_n)$

$\text{rlocfind}$

الدوائر التي  $\zeta + j\omega_n$  لها بعض خصائص damping +  $\omega_n$  يمكن استخدامها عند الحاجة

يعطي  $\text{rlocfind}$  gain و poles لأي نقطة محددة

يمكن ضبط هدف

Back to matlab to solve Ex 8.7 :

$z = 0.45$

$\omega_n = 0$

(a) شرح كود

$\gg \text{doc sgrid}$

$\gg \text{sgrid}(0.45, 0)$

يظهر الرسم التالي عند  $\zeta = 0.45$

له بعض خصائص  $\omega_n$  ولذا نستخدمه

$\gg \text{sgrid}(0.45, [0.1 \ 1])$

(b)  $[K, P] = \text{rlocfind}(G)$

نختار نقطة  $P$  في الرسم  $\rightarrow$  نكتب  $\text{rlocfind}$  بعد ذلك  $\rightarrow$  select a point in the plot  $\rightarrow$  نكتب gain و poles

(c)  $\text{rlocfind}$  command  $\rightarrow$  نكتب  $\text{rlocfind}$   $\rightarrow$  نكتب poles  $\rightarrow$  نكتب gain  $\rightarrow$  نكتب  $P$   $\rightarrow$  نكتب  $K$

Advance

$\rightarrow$  See 8.7: transient Response Design via Gain Adjustment.  
 $\rightarrow$  Assuming C.L Loop sys is approximated to a 2nd order sys.

See Figure 8.20 (p. 408):

نلاحظ في Figure 8.20 (b) poles  $\rightarrow$   $\text{Im} = \dots$

(b) has better approximation than (a)  $\rightarrow$   $P_3$   $\rightarrow$   $\text{Im} = \dots$

(d) has better = (c)  $\rightarrow$   $\text{Im} = \dots$

لأنه بعد طائرته  $\rightarrow$  gain  $\rightarrow$   $P_3$   $\rightarrow$   $\text{Im} = \dots$

check page 407 and 408

Lee 19:

How to use MATLAB to draw Root-Locus:

→ Ex 8.8 (p 408):-

→ Open loop  $F(s) = \frac{k(1.5+s)}{s(s+4)(s+10)} = \frac{k(1.5+s)}{s^3 + 11s^2 + 10s}$

1 → numgh = [ 1 1.5] //  $s^2 - 1.5s$  roots are 0 and 1.5

2 → dengh = poly([ 0 -1 -10 ]) //  $s^3 + 11s^2 + 10s$  roots are 0, -1, -10

OR 2 → dengh = [ 1 11 10 0 ] roots are 0, -1, -10

3 → G = TF(numgh dengh) // create a TF (G) and display it.

4 → rlocus(G) // Draw RL of function G.

5 → Sgrid(0.8 0); // To draw a line of specific value of  $\zeta$  and/or  $\omega_n$ , here  $\omega_n = 0$  coz I don't want to draw it. [از این  $\omega_n$  خطی را نمی‌کشیم]

6 → [Ki Pi] = rlocind(G); // to determine the gain (K) and C.L poles (p) for a selected point on the R-L.

گزارش پنجره گرافیک را ببینید → selected point x+yi

K // selected gain

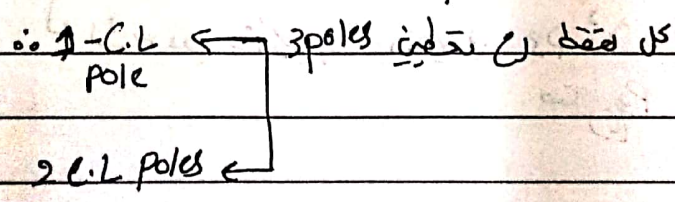
$K = 2.36$

p // pole

$p = -0.82 + 0.66j$

$= -0.82 + 0.66j$

$= 9.25 + 0.00j$



(Complex conjugate).

\*) جبر الیه می‌رسد که آنجا نقاط تقاطع  
\*) با افق و بطول خط تقاطع 3 poles تقریبی و همه منهای تقاطع است

approximation

Case	Third C.L. Pole	→ Cancellation
1	-9.25	-1.5
2	-8.61	
3	-1.8	

7>>  $T_i = \text{feedback}(k_i * G_{ol})$  // This command is used to find C.L. TF.

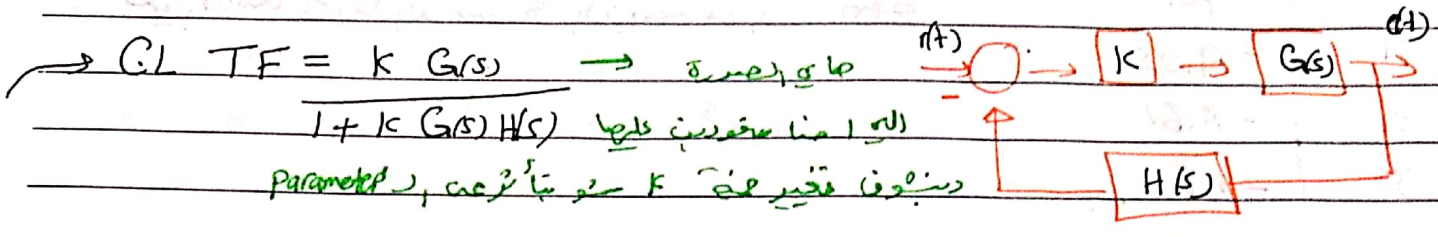
8>>  $T_i = \text{feedback}(1 * G_{ol})$   
 $k_i = 7.38$  → unity feedback

8>>  $\text{step}(T_i)$  ↓ // Generate closed-loop response for point select on root locus.  
 → generate this step response for the 3 points.

Note This example gives us the overshoot value (152%). we can find  $\zeta$  from the equation  $\%OS = e^{-\zeta\pi/\sqrt{1-\zeta^2}} * 100\%$   
 OR From figure 4.15 (page 126).

Lec 20:

Sec 8.8: Generalized Root-Locus (see Figure P 412)



If the function  $K(G(s)H(s))$  is formed as: (for example):-

Function of  $s$

$$K(G(s)H(s)) = \frac{10}{(s+2)(s+10)}$$

let the C.L. as this configuration

$$C.L. TF = \frac{10}{s^2 + 2s + 10}$$

Function of  $s$

Now  $K(G(s)H(s)) \equiv \frac{10}{s^2 + 2s + 10}$   $\neq$   $\frac{10}{s+2}$   $\therefore$  not they have same configuration

Sec 8.9 Root Locus positive - feedback systems:

-ve feedback  $\xrightarrow{sys}$   $TF = \frac{K G(s)}{1 + K G(s) H(s)} \rightarrow K G(s) H(s) = -1 = (k+1)18^\circ$

the =  $\xrightarrow{sys}$   $TF = \frac{K G(s)}{1 - K G(s) H(s)} \Rightarrow K G(s) H(s) = 1 = k(360)^\circ$

$k = 0, \pm 1, \pm 2, \dots$

Ex 8.9 p(414)

skip see 8.10

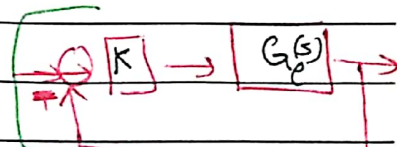
End of #8

# #9: Design via Root-Locus:

In general, increasing the value of  $K$  will reduce steady-state error, but overshoot will increase [assuming that C.L system is still stable].

Ex) step response error  $e(\infty) = \frac{1}{1+K_p}$

position error constant

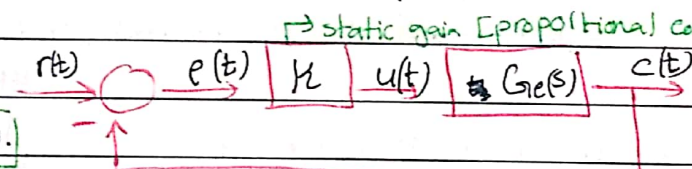


→ as  $K$  increases ↑,  $K_p$  will increase ↑  
 as a result  $e(\infty)$  will decrease ↓

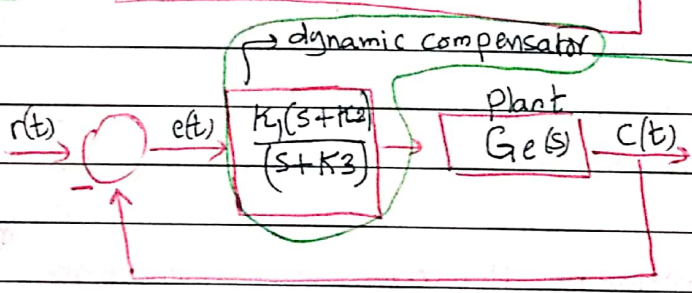
$$K_p = \lim_{s \rightarrow 0} K G(s) = K \lim_{s \rightarrow 0} G(s)$$

constant ex=1

Instead of using a static gain controller (proportional controller), we will use Feed Forward dynamic compensators (might contain zeros and/or poles in their TF).



⊗  $e(t) = r(t) - c(t)$



by using dynamic compensators we can improve both transient and SS error at the same time.

⊗ we use dynamic compensators in order to reach the desirable overshoot, % error etc values as much as possible.

لقد استخدمنا كبريت

⊗ : كبريت  
 (SS.e) ... overshoot ...



dyn Comp

Transient Response SS. error parameters overshoot,  $\times$   
 SS. error is independently  $\times$

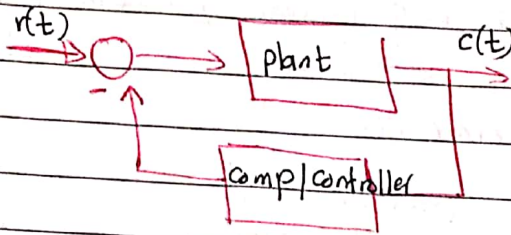
Lec 21:

Controller OR compensator  $\times$

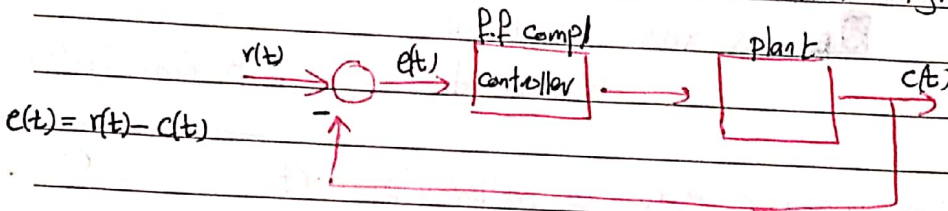
$\times$  We have few types of Compensator/Controller configurations:

1) Feedback compensator/Controller:

Feedback.



2) Feedforward compensators/Controller: takes error signal as an input

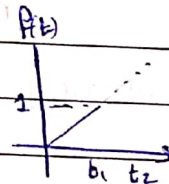


$\times$  we will study feedforward compensator design [Configuration] 2

$\times$  What effect would you expect on transient response, when error signal is differentiated in the compensator?

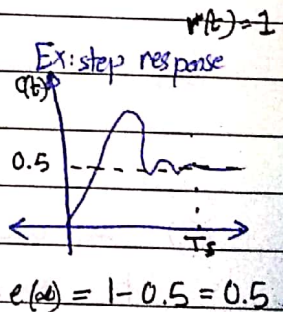


Differentiating the error is expected to reduce settling time and reduce the overshoot in transient response.



[that is, it will improve transient response (أحسن)]

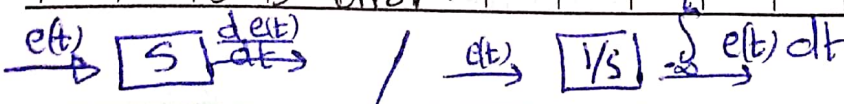
$\times$  Note that, differentiating the error signal will not affect (improve) the SS. error (But) integrating the error will improve the steady-state error.



Conclusion: - For Feedforward compensators,

differentiating  $e(t)$  will improve transient response ; however integrating  $e(t)$  will improve (reduce)

S.S. error.



قبل لأعمل design ل feed back comp. لازم نحدد ال unity ← sys  
 feed back ← sys ← unity

Lec 22:

(Integral Comp) مع ال SS error ↓

Sec 9.2: Improving SS error using integral compensation

Fig 9.3 page 454: Figure 9.3: (b) تغيير ال R-L ل تغيير ال TF ل النظام

In (a): انضنا (K) متنظ و مع ال unity ال error ال كادوب و لكن بيدي أقل من ال error

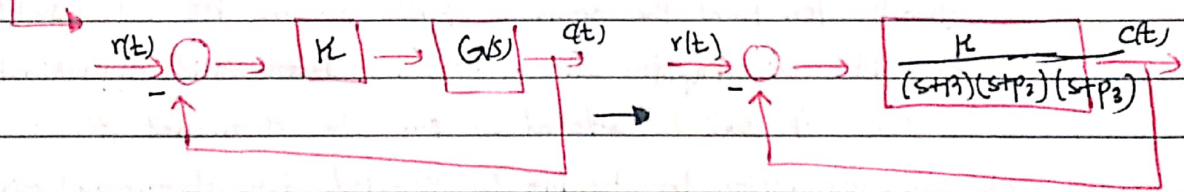
In (b): على الأقل من ال error انضنا ال error ال كادوب و لكن ال error ال كادوب

In (c): على الأقل من ال error ال كادوب ال error ال كادوب ال error ال كادوب

← ال زيادة ال open loop poles  
 ال T.F. ال تغيير ل ال T.R. ال تغيير  
 و ال unity ال R-L ال تغيير

← ال زيادة ال open loop poles  
 ال T.F. ال تغيير ل ال T.R. ال تغيير  
 ال error ال كادوب ال error ال كادوب ال error ال كادوب

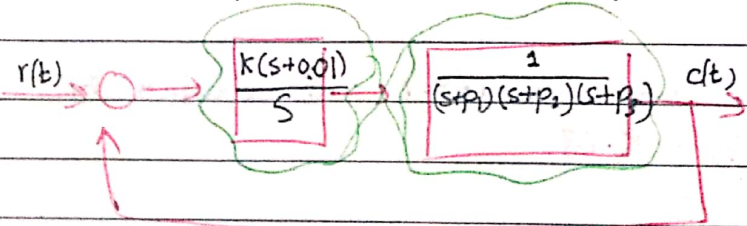
Fig. page 454:



system types Type 0

$K_p$  is finite,  $e_{ss} = \frac{1}{1+K_p}$

to get rid of S.S error, we increase system type through cascade compensation [Integral compensation].



System is Type 1

$K_p = \lim_{s \rightarrow 0} G(s)$

$K_p = \lim_{s \rightarrow 0} \frac{K(s+0.01)}{s} \frac{1}{(s+p_1)(s+p_2)(s+p_3)}$

$K_p = \infty, e_{ss} = \frac{1}{\infty} = 0$  Five Apple 0

⊗ If the system is type 0 and  $K_p$  is finite  $\rightarrow$  to eliminate  $e_{ss}$  we make the system type 1 by adding a pure integrator.

⊕ If the system = type 1 =  $K_v$  =  $\infty \rightarrow \infty$  we make the system type 2 by adding  $\frac{1}{s^2}$ .

Example 9.1 (p 455):

Given the root locus overshoot  $\zeta = 0.174$   $\rightarrow$  use matlab command `Sgrid` to get the line with constant damping ratio.

also use closed loop poles  $\rightarrow$  find the root-locus poles  $\rightarrow$  line  $\zeta = 0.174$

go to Ch 4 p: 176

for  $\zeta = 0.174 \rightarrow$  overshoot  $\approx 60\%$

Fig. 4.15

when  $K=0$  : c.l poles are the o.l poles

$\rightarrow K=164.6$  :  $-0.694 \pm j3.926$  2 complex conjugate poles  $\rightarrow$  dominant poles

$\rightarrow K=164.6$  :  $-0.694 \pm j3.926$  2 complex conjugate poles  $\rightarrow$  dominant poles

dominant poles

dominant poles

is  $\zeta = 0.174$

of line.

note dominant poles at  $p = -1, -2$

and pole at  $p = -10$  : The third pole is

almost ten times the dominant poles so the ~~the~~ T.R results from it will rapidly die. So, the ~~valid~~ valid approximation / real T.R will be affected by only the dominant poles.

$\hookrightarrow$  Second order approximation by taking for granted the dominant poles and neglect the third one at  $p = -10$ .

Q: Is it a good approximation? according to rule of thumb

If the real part of the Dominant poles is 5 times less than the third pole, then the approximation is a good one

$\hookrightarrow$  to find S.S error when  $K=164.6$  : ① Calculate  $K_p$

$$K_p = \lim_{s \rightarrow 0} K G_c(s) = 8.23$$

$$e_{ss} = \frac{1}{1+K_p} = \frac{1}{1+8.23} = 0.108 \rightarrow \text{error}$$

figure (b) type 2 zero

Figure (a): uncompensated

Figure (b): compensated

→ after calculating the error for figure (a) → go to figure (b) to eliminate the error and keep the T.F.

↓  
add pure integ

↓  
add a zero

\* after changing the O.L T.F to reach the same result at  $\gamma = 0.174$  the gain changed from 164.6 to 158.2 and the 2 complex conj

changed from  $-0.694 + j3.926$  to  $-0.688 \pm j3.832$  with the same Root-Locus (منه)

9.6 (b) تعديل

Note that at  $k = 158.2$ , the added pole (fourth pole) is at point  $p = -0.0902$

[بالتقريب] very close to the added zero at  $z = 0.1$  [بالتقريب]   
 وهو قريب من الصفر المضاف

\* Now It's a fourth order system but can be approximated casily the third pole at  $p = -10$  [بالتقريب]   
 قبل

↳ the fourth pole [بالتقريب]   
 زي ما ذكرنا فوق

dominant poles (القطب المسيطرة)   
 complex 2 poles (قطب معقدة)

\* Go to Fig 9.2: step response

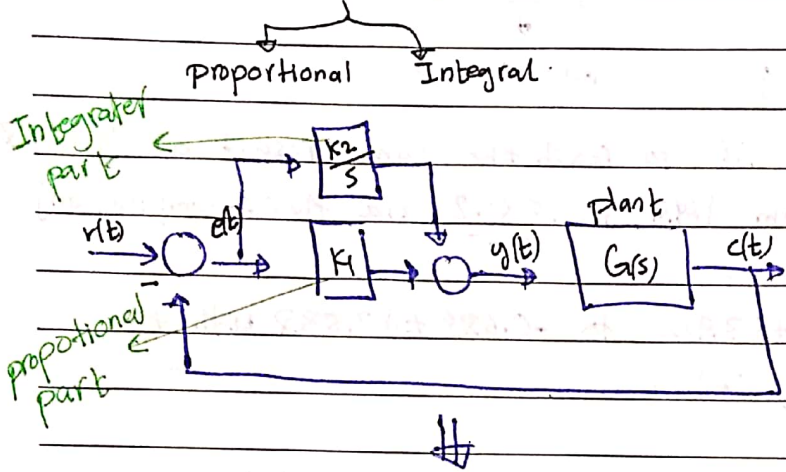
uncompensated response (استجابة غير معوضة) و compensated response (استجابة معوضة)

$e(\infty) = r(t) - ct = 0.108$  uncomp input  $\therefore 1 =$  desired input  $\leftarrow$  step input

$e(\infty) = 0 \rightarrow r(t) = ct = 1$   $\leftarrow$  compe input  $\therefore$

\* Both have same overshoot and similar settling time

\* Ideal Integral cascade compensator  $\left(\frac{K(s+a)}{s}\right)$ , is also called a PI-controller

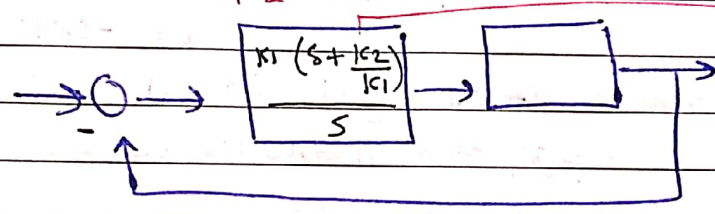


$$Y(s) = E(s) \left[ K_1 + \frac{K_2}{s} \right]$$

$$\frac{Y(s)}{E(s)} = \left[ K_1 + \frac{K_2}{s} \right]$$

$$= K_1 \left( 1 + \frac{K_2}{s K_1} \right)$$

Ideal PI controller

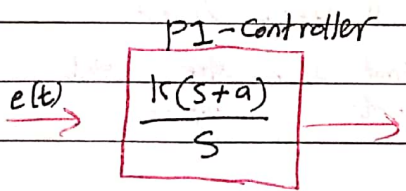


ex: 1022/20

$$\left[ \frac{15}{158(s+0.1)} \right]$$

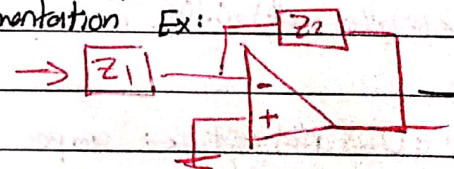
Lec 23: Ideal PI compensator includes pure integration (s in the denominator of compensator T.F)

\* To implement (realize) a PI compensator, an active element is required such as an Op-Amp. (PI-controller)



[circuit diagram of PI controller] (\*)

implementation Ex:



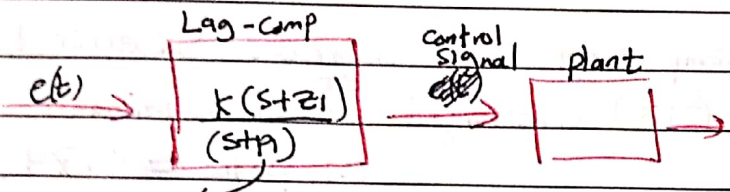
(active element) operational Amplifier

External P.S. Power source

Requires external power

Alternative, نوع من

Alternative to using OP-Amp (active element) in realizing a PI-controller is to use something called a Lag compensator [non-ideal PI-controller]

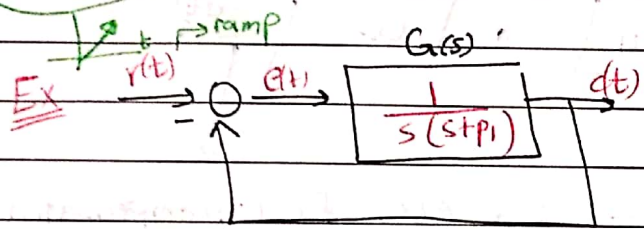


freq. response كى كى phase delay كى كى pure integrator كى كى

does not have pure integrator

∴ It doesn't increase system type.

go to book page 458, see Figure 9.9



∴  $K_U$  is finite (non-zero value)

S.S.e =  $e(\infty)$  for a ramp input

$$e(\infty) = \frac{1}{K_U} = \text{finite (non-zero value)}$$

$K_U = \lim_{s \rightarrow 0} s G(s)$ , In general increasing the value

of any type error constant ( $K_p, K_U, K_a$ ), decreases  $e(\infty)$ .

from page 459:

with Lag-comp  $K_{new} = \frac{K_z z_c}{p_c} > K$ ,  $e(\infty) \propto \frac{1}{K_U}$

without

Lag Comp, كى كى

error constant كى كى

كى كى

then:  $e(\infty)_{new} < e(\infty)$

كى كى

كى كى

كى كى

ideal comp, كى كى

See Figure 9.10 p: 459:

T.R, كى كى  
 R-L كى كى  
 Lag comp كى كى  
 كى كى

كى كى

⊗ Note: For the Lag Compensator the added zero and pole

$$z > p$$

in this ex  $z = 0.111 > p = 0.01$

29

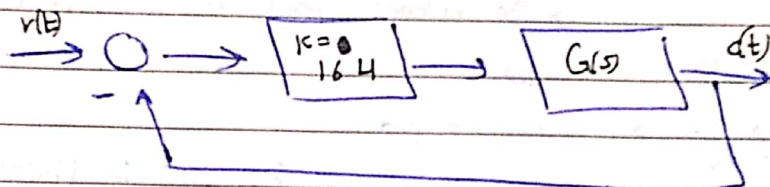
Ex 9.2 Lag Compensator design for type 0 system:-

Notes: For type zero system increasing the static error cons. ( $K_p$ ) will decrease SS error.

Goal: is to design a lag compensator that decreases S.S. error, but maintains transient response (T.R) unchanged [  $\zeta$  of dominant C.L poles = 0.174 ].

We started with:

⊗ Uncompensated system with C.L poles having  $\zeta = 0.174$



⊗ For type 0 sys:  $e(\infty) = \frac{1}{1+K_p}$  →  $e(\infty)$  For Uncompensated sys

$$\rightarrow K_{p_{old}} = \lim_{s \rightarrow 0} K G(s) = \lim_{s \rightarrow 0} \frac{16.4 \times 1}{(s+1)(s+2)(s+10)} = 8.23$$

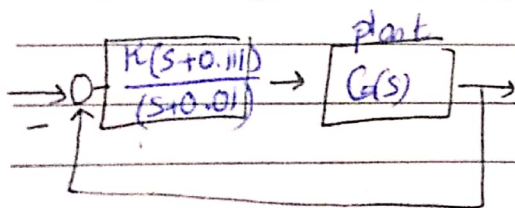
$$= \frac{1}{1+K_{p_{old}}} = \frac{1}{1+8.2} = 0.108$$

now to reduce the error by (10) times  $e(\infty)_{new} = 0.0108$

⊗ After using lag compensator, we want  $e(\infty)_{new} = 0.0108$

$$e(\infty) = 0.0108 = \frac{1}{1+K_{p_{new}}} \text{ then, } K_{p_{new}} = 91.6$$

compensated system using lag comp



⊗ كمان نريد ان نضع pole جديد

قريب من  $\text{Im} = -\sigma$  من الاصل

then

$$p = 0.01$$

من اجل ان نضع zero جديد

$$z_c = \frac{K_{p_{new}}}{K_{p_{old}}} = \frac{91.6}{8.23} = 11.13$$

من اجل ان نضع pole جديد

$$K_{p_{old}} = 8.23$$

⊗ check Figure 9.13 page 461

⊗ for lead compensators

$$P > Z$$

Lec 24:

OP-AMP, etc

Check 9.6 section, page 495 + 496 + 498 → table 9.11 (Lag Comp)  
↳ table 9.10

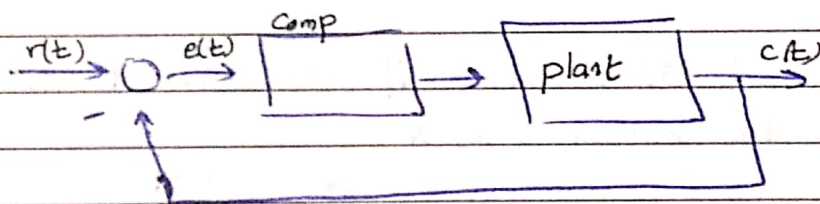
⊗ Note from table 9.11 T.F for the Lag Comp:

$$K \frac{R_2}{R_1 + R_2} \frac{s + \frac{1}{R_2 C}}{s + \frac{1}{(R_1 + R_2) C}}$$

$P_1$

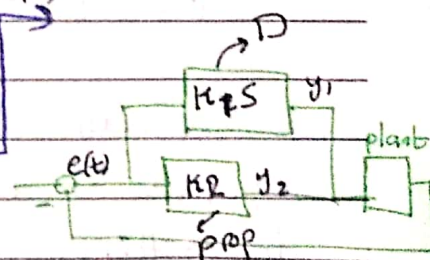
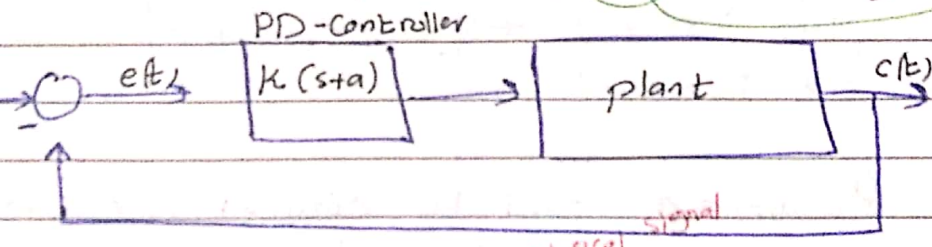
⊗ From previous Lec  
Lag Comp was:  
$$K \frac{(s + Z)}{(s + P_1)}$$

Sec 9.3: Improving Transient Response:  
via Cascade Comp.



⊗ Improving the T.R usually means decreasing settling time and decreasing rise time  $T_r$ , while maintaining a small overshoot.

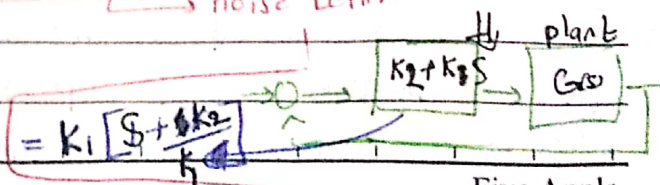
⊗ Ideal derivative compensator (PD-controller) → gain + zero



⊗ Note: if the error =  $\sin(t) + 0.1 \sin(27.50 t)$

→  $e(t) \approx \sin(t) + 0.1 \sin 314t$

→  $\frac{de(t)}{dt} \approx \cos(t) + 31.4 \cos 314t$



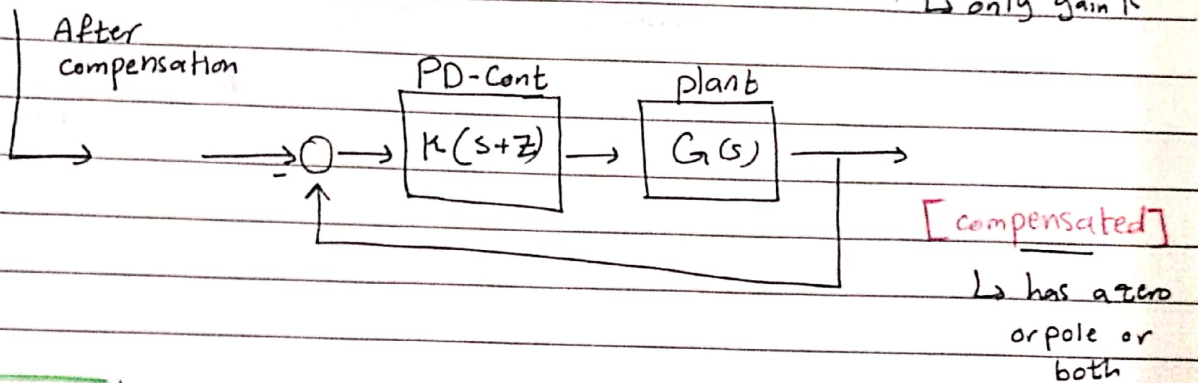
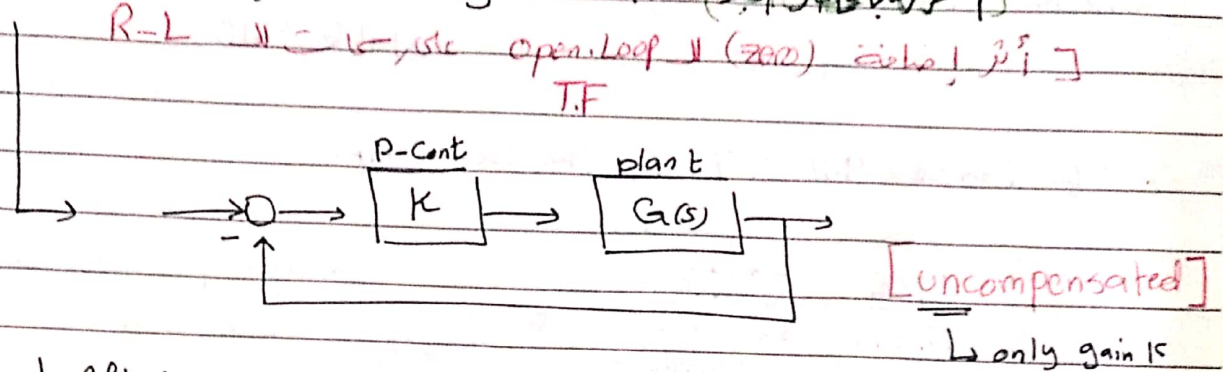
∴ differentiation can lead to amplifying the noise (high freq. component)

Five Apple Component



⊗ Before using a PD-Controller we have to check it first; Because sometimes with simulation it may give us a good response but when using it for real the response may be very different due to noise appearance [PD-Cont Amplify the noise]. 31

⊗ Check page 464, Figure 9.15 (فحص صفحة 464، الشكل 9.15)



Lec 25: ⊗ check (25) video if you want to find the matlab files ⊗

→ Check page 464 Figure 9.15

⊗ الفكرة هي عندنا برامجة [uncompensated] (a) يعني متك منظم [gain K] sys

يعني نرفع b, c, d ونضيف ٠ (zeros) على a (مفيدة)

Table 9.2 → عمل هذا الأمر بالبرامجة + [نصفنا ترونا على برامجة] حيث هو ٠.٤ (٠.٤) في الحالة = 0.4

For (a) and (b) → ① adding a zero = -2 to be results: increasing the magnitude of the real part from |0.939| to |3| which causes the settling to decrease ② overshoot is fixed for all figures

③ K increased → from K = 23.92 to 51.25

④ wd increased so the peak time decreased

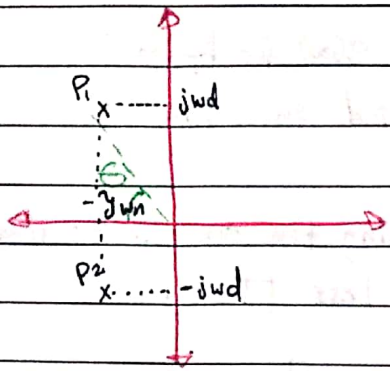
⑤ since  $\zeta = 0.4$  is fixed and wn increased then Tr decreased

③  $\cos\theta = \zeta$   
 $\theta \propto \zeta$

④ Ch 4, Figure 4.15  
 page 176  
pp

T.R.  $\zeta$  is Dominant Poles  
 Parameters

Recall / For an underdamped 2nd order system  
 that  $0 < \zeta < 1$



$$T(s) = \frac{1}{(s-p_1)(s-p_2)}$$

$$(s-p_1)(s-p_2) = 0$$

roots  $s_{1,2} = p_1, p_2$

$$p_1 = -\zeta\omega_n + j\omega_d$$

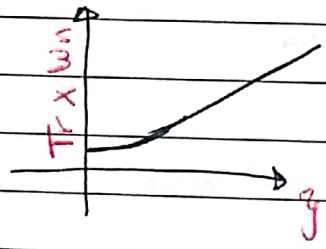
$$p_2 = -\zeta\omega_n - j\omega_d$$

where  $\omega_d = \omega_n\sqrt{1-\zeta^2}$

$T(s) = \frac{4}{\zeta\omega_n} = 4\tau \Rightarrow \tau \propto \frac{1}{\zeta\omega_n}$

$\Rightarrow T_p = \frac{\pi}{\omega_d} \Rightarrow T_p \propto \frac{1}{\omega_d} \Rightarrow T_p \propto \frac{1}{\text{Im}[p_1, p_2]}$

$\Rightarrow Tr \Rightarrow$  Check Ch 4, page 177, Figure 4.16 (pp)



اذا ثبتنا  $\omega_n$  و زدنا  $\zeta$   $\rightarrow$   $Tr$  کمتر  
 کمتر  $\zeta$   $\rightarrow$   $Tr$  زیادتر  
 اگر  $\zeta$  ثابت باشد و  $\omega_n$  زیادتر شود  $\rightarrow$   $Tr$  کمتر  
 اگر  $\omega_n$  ثابت باشد و  $\zeta$  زیادتر شود  $\rightarrow$   $Tr$  زیادتر

Ratio  $\omega_n \times Tr$  constant  
 ع.ل  $\omega_n \times Tr$  constant

- Conclusion:
- ① If  $\zeta$  is Fixed, increasing  $\omega_n$  will decrease  $Tr$ .
  - ② If  $\omega_n$  is Fixed, increasing  $\zeta$  will increase  $Tr$ .

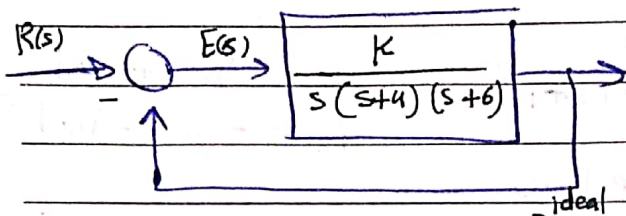
Hint

Note: If  $T_p$  decreased, the  $Tr$  decreased [as an Indication]

⊗ For overshoot = 0.16  $\rightarrow \gamma = 0.504$

Lec 26:

Ex 9.3: PD-Controller design, Page 466:

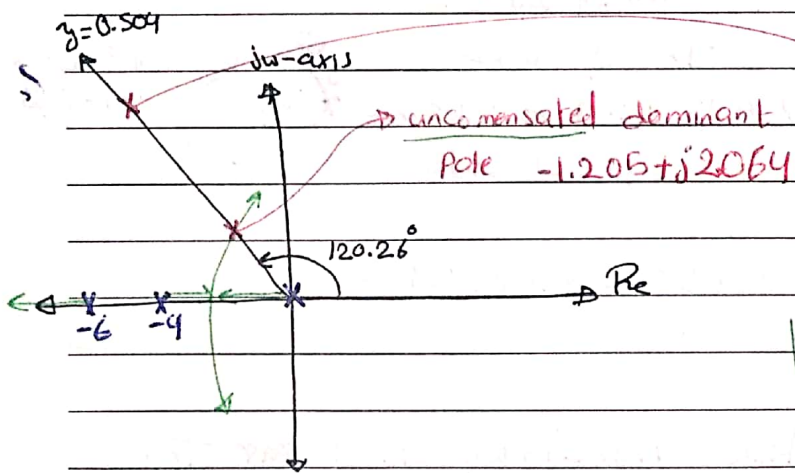


$\rightarrow$  type 1;  $e(\infty)$  for  $k_p = 0$   
 $\rightarrow$  uncompensated system.

Goal of ⓐ: design a PD-Controller, remaining the overshoot value of 0.16 and at the same time get a 3-times less ( $T_s$ ).

$$\rightarrow T_{s_{new}} = \frac{1}{3} T_{s_{old}} \quad \therefore \begin{bmatrix} T(s) \propto \frac{1}{\zeta \omega_n} \\ T(s) \propto \frac{1}{\text{Re}[Pole]} \end{bmatrix}$$

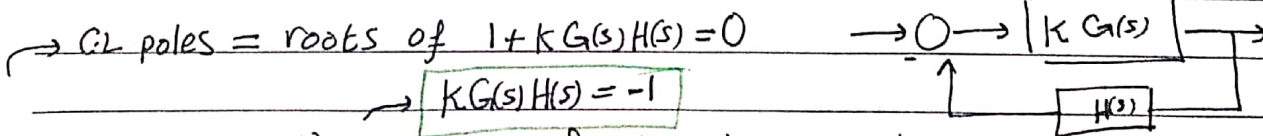
Ex If uncompensated system has  $(\zeta \omega_n)_{old} = 1$ , then the required compensated system has to have dominant poles with real part  $\text{Re}[Pole] = -\zeta \omega_n = -3 \quad \therefore (\zeta \omega_n)_{new} = 3(\zeta \omega_n)_{old}$



desired compensated dominant pole  
 $-3.613 + j6.193$

$\rightarrow$  We have to use PD-Controller with a zero value that will give us the desired pole value. will keeping a fixed value of  $\zeta$ .

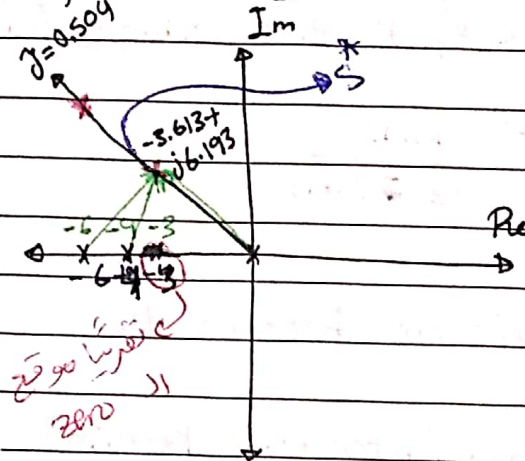
TF of:  
 O.L =  $G(s)H(s)$   
 C.L =  $\frac{K G}{1 + K G H}$



Any point(s) that satisfy the phase condition:  
 $\angle G(s)H(s) = \angle (K+1) 180^\circ$ ,  $K = 0, \pm 1, \pm 2, \dots$  will lie on R-L.

Recall:  $\angle F(s) = \left( \frac{s+z}{s+p} \right) = \angle(s+z) - \angle(s+p)$

Finding the value of the added zero for the PD-controller.



$\angle G(s^*) H(s^*) = (2K+1) 180^\circ$

to find the phase, we draw a vector from each pole and find its angle

$\therefore \angle(s^*)_{\text{uncompensated}} = \sum \theta_z - \sum \theta_p$

$= -(\theta_1 + \theta_2 + \theta_3) \approx -275$

$p_1 = -3 \leftarrow$   
 $p_2 = -4 \leftarrow$   
 $p_3 = -6 \leftarrow$

$-275 + \phi_{\text{comp}} = -180$

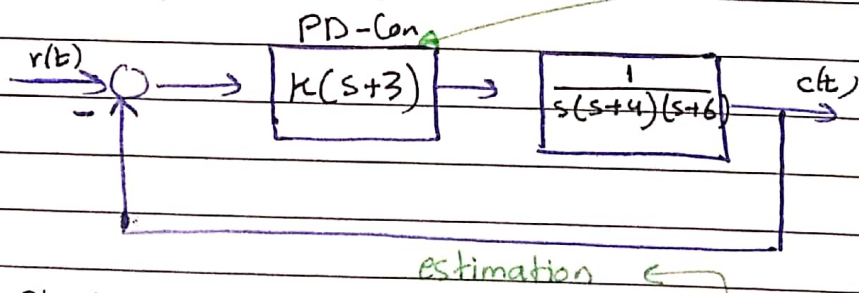
the nearest valid point to -275

according to  $(2K+1) 180^\circ$  vs  $(2K+1) 180^\circ$  is RL

$\phi_c \in [0, 180]$

$\therefore \phi_c = 95^\circ$  [the added zero will have angle of  $95^\circ$ ]  
and will be located at point -3

Compensated system now:



$\otimes$   $K_c$  changed from 43.35 to 47.45 (From MATLAB)

$\otimes$   $T(s)$  changed from 3.32 to 1.107

$\otimes$  Check table 9.3 page 467: the uncompensated and compensated values are taken according to the two dominant poles [2nd order approximation]  
 $\rightarrow$  And the simulation values are taken according to all three poles [since it's a third order system].

(actual values)

$\otimes$   $e(\infty)$  decreased [error for a ramp input]

$\otimes$  Also, note that the TP decreases because  $\omega_n$  increases when  $\gamma$  is constant.

$\rightarrow e(\infty)_{Kp} = 0$   
 $\rightarrow e(\infty)_{Kv} = \frac{1}{Kv}$   
 $\rightarrow e(\infty)_{Ka} = \infty$

\* How to calculate the exact value of the zero using  $\phi_c$ ?

Sol:

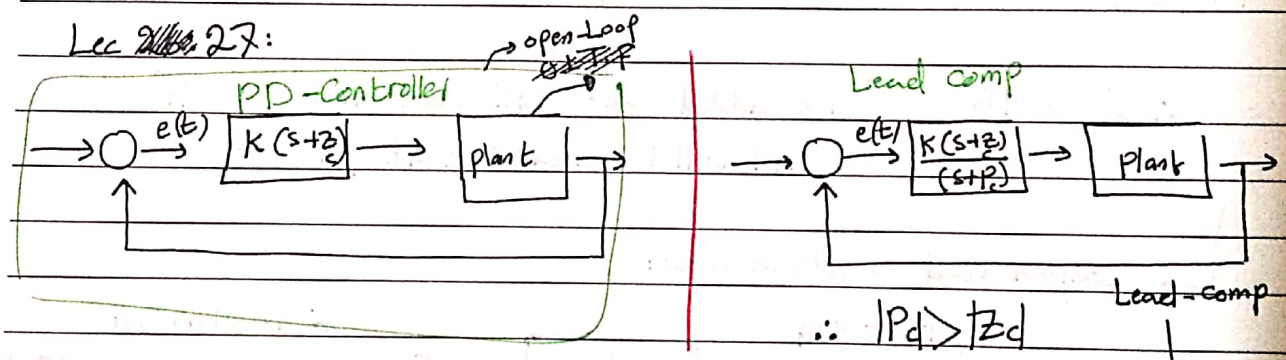
$$\angle (s+z_c) \Big|_s^* = 95^\circ = \tan^{-1} \frac{6.193}{z_c + 3.613}$$

Check Figure 9.22, page 469: the difference between the uncomp. and comp. response.

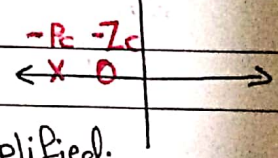
[note that the comp. sys is much faster while keeping] a constant overshoot and a  $e(\infty)$  of zero

system type 1, step input

Lec 27:



$\therefore |p_c| > |z_c|$



Advantages For Lead comp:

If there was any high-freq noise, It won't be amplified.

Disadvantages For Lead comp:

Can't increase the # of zeros in the O.L.T.F.

(But The PD-controller adds a zero to the O.L.T.F.)

infinite  $\rightarrow$  zero  $\rightarrow$   $\infty$

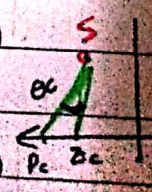
Check Figure 9.24, page 420

\* بدي اكل design ل  $\theta_c$  بحيث اقدر اقل  $\zeta_c$  و  $p_c$  بحيث يكونوا مرتبين

بجانب من طريقه  $\theta_c$

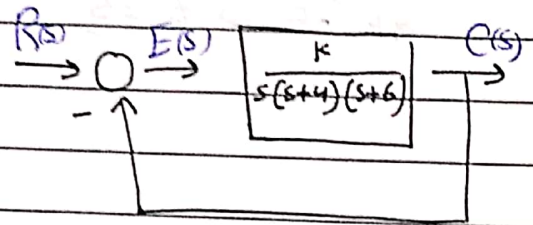
\* يعني بقدر اقدر ل  $\zeta_c$  و  $p_c$  اري  $p_c$  بناه اس  $\theta_c$  بحيث

الماخذ  $\theta_c$  فبتسايفهم كالماتار هو اقلهم



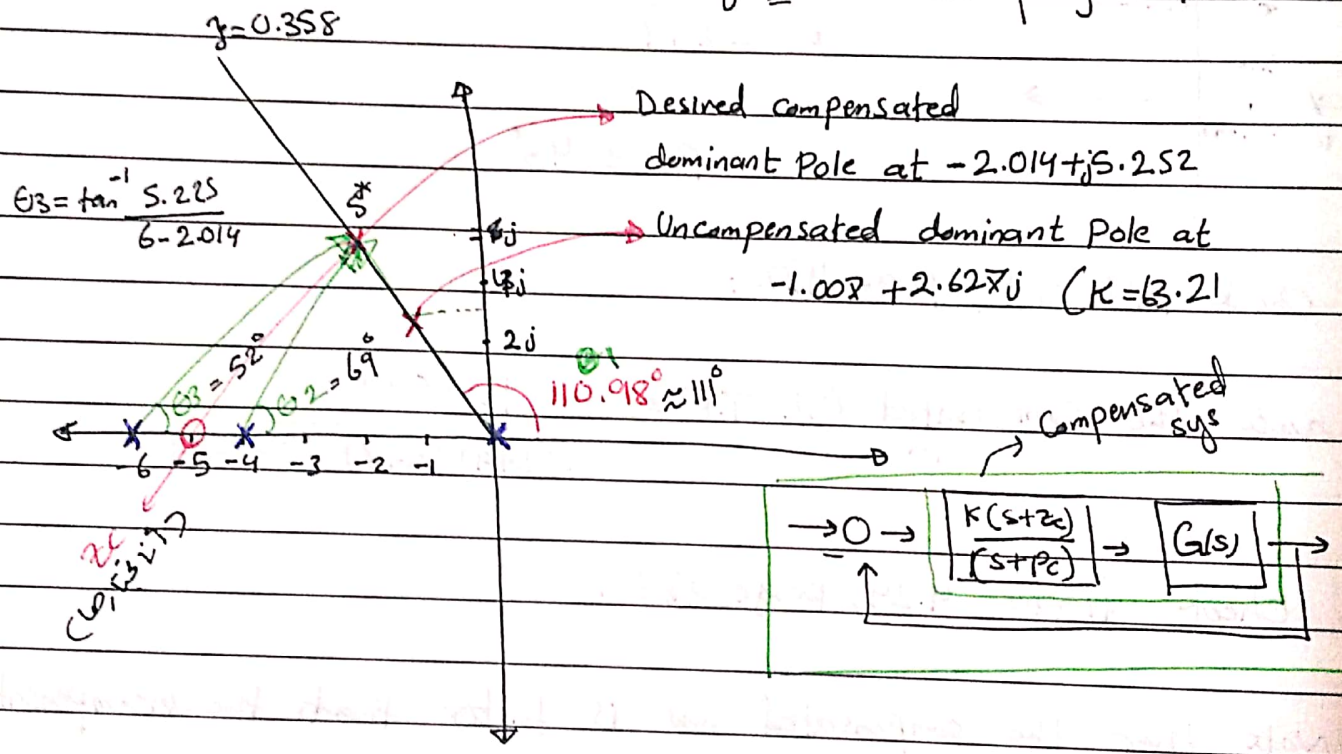
Ex 9.4 page 471: Check Figure 9.26

For this system  $\downarrow$



Goals: Desired OS% = 30%  
 From table ( $\zeta = 0.358$ )  
 Figure

then design Lead-Comp: reduce the  $T_s$  by  $\frac{1}{2}$  while keeping 30% overshoot



$\rightarrow T_{s_{new}} \propto \frac{1}{\zeta \omega_n}$  then  $(\zeta \omega_n)_{new} = 2 (\zeta \omega_n)_{old}$

$(\zeta \omega_n)_{new} = 2 |Re \{ \text{Dominant Poles} \}| = 2 * 1.008 = 2.014$  (RL  $\downarrow$ , overshoot  $\downarrow$ )  
 RL  $\downarrow$ , overshoot  $\downarrow$  (angle)  $\downarrow$   $\rightarrow$   $\zeta \omega_n$   $\downarrow$

$\rightarrow -\theta_1 - \theta_2 - \theta_3 = -232.4^\circ$

$-\theta_1 - \theta_2 - \theta_3 + \theta_c = (2k+1) 180^\circ \therefore \theta_c = \theta_{z_c} - \theta_{p_c}$   
 $\theta_c \in [0, 180^\circ]$

$\rightarrow -232.4^\circ + 52.4^\circ = -180^\circ$  (From this)

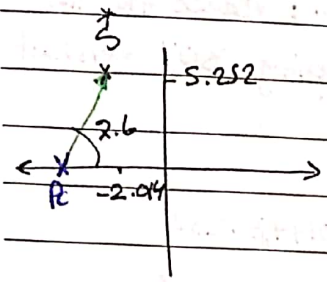
⊗ الأيمن والأعلى أحط أن zero من Loop O.L. زي عند (-6) لأن من بين بعض cancellation لـ zero فلا أضيف pole ما يكون أثره في poles النهائي

→  $\theta_c = 52.4 = \theta_{zc} - \theta_{pc}$  , let compensator zero

∴  $\theta_{zc} = \tan^{-1} \frac{5.25}{5-2.014} \approx 60^\circ$

be [at -6 →  $z_c = 6$ ] <sup>جدد</sup>  
 But following the book  
 at (-5 →  $z_c = 5$ )

→  $\theta_c = 52.4 = 60^\circ - \theta_{pc}$   
 $\theta_{pc} = 7.6^\circ \rightarrow$  Find  $P_c$



$$\frac{5.252}{P_c - 2.14} = \tan 7.6$$

$P_c \approx 43^\circ$

Check table 9.4, page 482 :

note the compensated O.L.T.F =  $\frac{K(s+5)}{s(s+4)(s+6)(s+43)}$

check figure 9.29, page 483 :

↳ Note that the compensated sys is faster than the uncompensated sys. (Tr and Ts decreased)

↳ Note that for system C, the overshoot is not 30% so it was a little off the sys needs, the design were actually done for 30% OS but after simulation the actual OS was 14.5 [this error is due to the assuming that it is 2nd order though it's not]

∴ Because of 2nd order approximation.

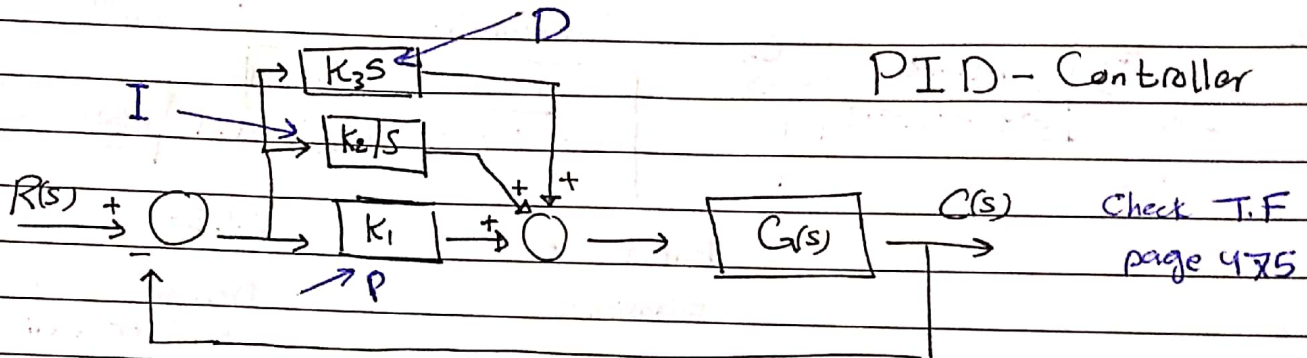
↳ Note that the error is reduced too.

Conclusion: Lead-compensator affect the transient response to reach the desired one, and Also can reduce the SS. error.  
 لا بد من التعديل بالجزءة معينة

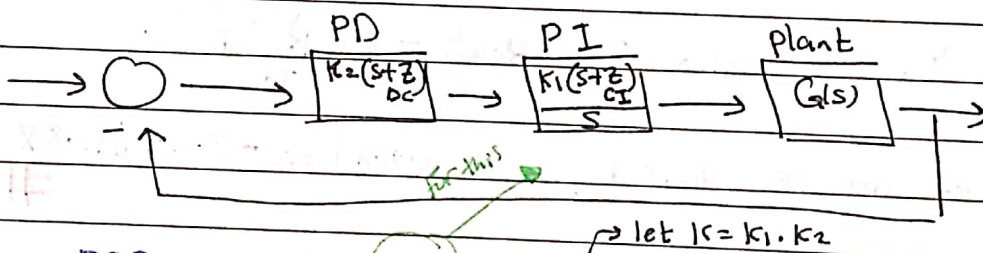
Lec 28:

9.4 PID and Lead-Lag compensator:

Figure 9.3, Page 475:



|| equivalent to [ have PD and PI controllers in series ]



PID-Controller T.F = 
$$\frac{k (s+z_{pc})(s+z_{zc})}{s}$$

Design Process steps: (for PID-Controller)

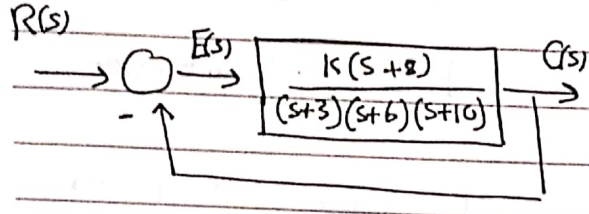
- 1) Design PD-Controller to improve the T.R.
- 2) Simulation [to check that we got the desired response] T.R
- 3) Design PI-controller (add a pole at origin) and add a zero near the pole
- 4) Simulation again (check the steady-state error) good or not?

IF not: re-design again, change the zeros positions and check again.



Example 9.5, page 475

For this system:



Goal: Design a PID-Controller

For 0-type sys

In which:

$T_{p_{new}} = \frac{2}{3} T_{p_{old}}$   
 at OS% of 20%

$e(\infty) = 0$

desired transient

$\zeta = 0.456$

Desired  $e(\infty)$  for a unit step

$T_p \propto \frac{1}{\omega_d} \Rightarrow T_p \propto \frac{1}{\text{Im}\{\text{pole}\}} \Rightarrow \frac{2}{3} T_p \propto \frac{1}{\frac{3}{2} \text{Im}\{\text{pole}\}}$

to maintain  $\zeta$  value, also  $\text{Re}\{\text{pole}_{new}\} = \frac{3}{2} \text{Re}\{\text{pole}_{old}\}$

[these are the desired pole location] =  $-8.13 + j15.87$  # s

Now, the desired point is not on R-L, to make it the Root-locus:

We add a zero in which the angle of it will make the point on the R-L.

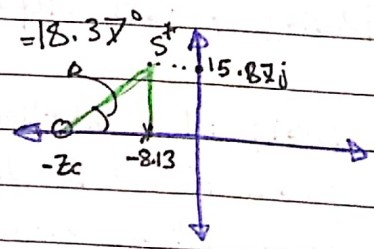
PD

→ zero

three pole

For un-comp sys

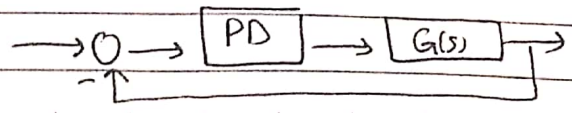
1) Sol:  $\theta_1 - \theta_2 - \theta_3 = -198.37^\circ$   
 $198.37^\circ - 180 = 18.37 = \theta_{DC}$



to find  $z_c$ :  $\frac{15.87}{z_c - 8.13} = \tan 18.37^\circ$

$z_c = 55.92$

∴ PD-Controller is :  $K(s + 55.92)$



Recall: PD-Controller may amplify the noise which is bad.

PD and PI require using OP-Amps (active elements)

Ho

PI

2) Sol:

$$G_{PI}(s) = \frac{K(s+0.5)}{s}$$

∴  $z_{IC} = 0.5$

∴ pole at origin, zero at  $s = -0.5$

Check Figure 9.35 :

$K=4.6$  ← For PID: faster transient response &  $e(\infty) = 0$

$K=121.5$  ← For uncomp: slower with  $e(\infty) = \text{Finite } \neq 0$

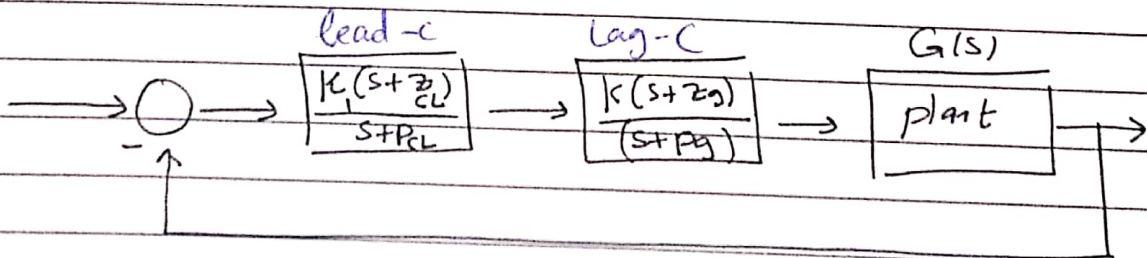
$K=5.34$  ← For PD: gives almost the desired response, but has an error

Check table 9.5, page [see the values difference]

↳ In Case I don't have any active elements OR I don't want to use PID-controller (PI, PD), In Stead we can use:

LAG-LEAD comp.

⊗ LAG-Lead compensator Design: (in Series)



→ Improve TR

steps: 1) start with the Lead-Compensator

2) simulation (check)

Reduce  $e(\infty)$  + 3) Lag-Comp design (note that the pole of it is not in the origin)

4) simulation IF good or not?

If not: design again (start with changing  $p_L$  and  $z_L$ )

Ex 9.6 [H.W]